

AVIATION WEEK

A MCGRAW-HILL PUBLICATION

NOV. 3, 1952

50 CENTS



FRIENDLY ENEMIES

One of the Navy's GRUMMAN GUARDIANS makes a pass over one of the Navy's submarines. It's a case of "friendly enemies" . . . for as the mongoose is trained to kill cobras, these big, carrier-based aircraft are designed to find and destroy submarines. One type of GUARDIAN, equipped with long range radar devices, hunts down the enemy. Then others, lighter on radar but heavier on bombs, come in for the "kill."

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Contractors to the Armed Forces

SUNDSTRAND'S CONSTANT SPEED DRIVES NOW ON GUIDED MISSILES!



More electronic devices on missiles accentuate need for AC power

For two good reasons Sundstrand Constant Speed Drives are now being applied to new types of guided missiles and are being considered for others now in drawing boards. These missiles carry even more electronic equipment than some of our new conventional aircraft; hence have a greater need for the weight and space savings of constant frequency AC power. Secondly, to assure delivery of the current to the target the guidance systems and other electronics devices require a reliable source of closely controlled constant frequency AC power. Sundstrand's quickly developed constant frequency AC power. Sundstrand's quickly developed constant frequency AC power. Sundstrand's quickly developed constant frequency AC power.



SUNDSTRAND AIRCRAFT HYDRAULICS

ISAACSTRAND MATCHING TOOL CO.
HYDRAULIC DIVISION, ROCKFORD, ILL.



Looks like a carpet, cleans like a dish

KEEPING a commercial airplane's carpet clean used to run into agony. The wool carpeting used by airlines sagged, dirt, soaked up stains and got gray like. Whenever it needed dry cleaning, which was often, it had to be removed from the plane. Extra carpets had to be kept on hand for quick replacement. And any solution to the problem which involved beauty of the cabin was unacceptable.

Then B. F. Goodrich engineers fixed up their Avonite flight rug. They developed a process of embow-

ing colored felt with crystal clear Avonite flexible material. The color and pattern possibilities provided by the new method are practically limitless. With a sponge backing, the comfortable, cushiony feel of rich carpeting is kept.

Besides, the new kind of flight rug for airlines when laid, it resists soot and scratches. It can't be torn by grease, oil, or any ordinary stain and chemicals. Things that are spilled on it don't sink in, can be easily wiped up. Thorough cleaning is done with soap and

water, without removing the rug from the plane.

Write for samples of the new Avonite flight rug material. Other BFG products for airlines include: seats, wheels and bodies, bonded rubber, De-luxe, Floor lock adhesives, Pressure Sealing Zipper, ball rolls, Rivets, accessories. The B. F. Goodrich Company, Avonite Division, Akron, Ohio.

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Forgings for the aircraft industry today demand the utmost in engineering and production techniques and in scientific laboratory control. This massive complicated landing gear component, weighing over 400 pounds, is typical of Wyman-Gordon's forging contribution to the ever-growing progress in aircraft design. In crankshafts for the automotive industry and in all types of aircraft forgings, steel and light alloy, Wyman-Gordon has pioneered in the development of forging "know-how"—there is no substitute for Wyman-Gordon experience.

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NEWS DIGEST

Domestic

Deagle XA-11F prototype two-seat carrier-based bomber has made its first flight of 30-min duration, at Edwards AFB, Calif.

The Mack Memorial Trophy for 1957, awarded annually by Royal New Zealand Aero Club, was presented to two American engineers, John E. Lundberg, Jr., and James W. Whelan, for their merit development of the Electronic Super Engine Assistant, at a luncheon given by the Institute of the Aeronautical Sciences in New York. The award commemorates Capt. Edwin C. Mack, who, with six companions, was lost on first commercial flight from U.S. to New Zealand in 1918, and is awarded yearly to the group or individual making most effective contribution to aviation safety. That is the first presentation of the award since 1940.

All-American Airways stockholders voted 100,000 to \$100 lowering split of the company into All-American Engineering Research Corp. with former AAA president Charles Wessell as president and treasurer and Allway Air Lines with Robert M. Love chairman of the board. They hope for CAB approval of the split by Jan. 1.

Maj. Gen. George W. Mendenhall, assistant deputy commanding general of Air Materiel Command, has been appointed to AFHQ as director of supply and services.

Associated Aviation Underwriters have issued their international airline trip insurance policy to provide minimum coverage of \$50,000 in case of death, according to D. R. Schmitt, manager Policy Unit. Loss has been increased from \$25,000 to \$50,000.

Air Force officially confirmed Aviation Week's report that General Curtis E. LeMay would remain as commander of the Strategic Air Force instead of succeeding General Nathan Twining in USAF vice chief of staff (AVIATION WEEK Aug. 25, p. 38). USAF and action to return General Twining in the post was taken by AF Secretary Fiedler with concurrence of Sen. Richard Russell, chairman of Senate Armed Services Committee.

Wiggins Airways has been cited for Dec. 31 because by a 12 majority decision of CAB. Northwest Airlines will take over some routes, Midwest Airlines others, while weaker points lose



BOGE SINDERSKY, dean of American Bell night dispatches, receives the annual trophy of the National Defense Transportation Assn. from USAF Secretary Thomas F. Fletter tonight in Washington, D. C. In making the award for being "the person contributing most to the field of transportation," Secretary Fletter called the various designer and builder "Mr. Helicopter."

on service altogether. Murphy Award Wiggins percent and future legions, while the minority and lost CAB route assignments made it that way.

Southwest Airways, Pacific Coast local service carrier, has purchased four 40-passenger Martin 202s. SWA also has two DC-7s.

Financial

United Air Lines reports record revenue for first nine months of 1957, with net company after taxes of \$8,267,000 of the \$4,460,000 represents net earnings after taxes for the third quarter alone. Operating revenues for that quarter amounted to \$45,176,000. United declared a quarterly dividend of 75 cents per share and an extra dividend of 15 cents per share on common stock, payable Dec. 15 to stockholders of record Nov. 14.

Republic Aviation Corp. showed a net profit of \$2,567,976 for the quarter ending Sept. 30, compared with \$792,875 a year ago. For the first nine months of 1957 gross sales totaled \$215,197,794, compared with \$81,077,497 last year. Unfilled orders at end of September were more than \$1 billion.

Norfolk Airlines reports per profit after taxes of \$982,634 for first three quarters of 1957, of which \$719,275 represented net profit for September.

Norfolk Aircraft shows a net in

come after taxes for the year ended July 31 of \$2,423,601 net of sales and other income, of \$187,450,926. Net income for the preceding year was \$1,276,053. Company's backlog of orders at close of fiscal year was \$495 million.

Ryan Aeronautical Co. has declared a regular quarterly dividend of 10 cents per share and an extra dividend of 10 cents per share, payable Dec. 12 to stockholders of record Nov. 21.

California Eastern Airways report a total net profit after taxes of \$944,595 for period ending June 30, compared to a net of \$82,004 for corresponding period last year. This does not include \$25,034 earned by Land-Air, Inc., a wholly owned subsidiary. A breakdown of the profits shows that company's operations resulted in net profits of \$533,113, \$315,154 profit from capital asset sale.

Delta Air Lines has declared a dividend of 25 cents per share, payable on Dec. 1 to stockholders of record on Nov. 14. Capital stock has been increased to \$5 million.

International

A British Overseas Airways Corp. jet Comet made a belly landing at Rome August Oct. 26 shortly after taking off for Johannesburg, South Africa, the first major accident in which a Comet has been involved. Nine of the crew or passengers were injured. An engine is reported to have failed.

Dassault Mystere 2 has attacked Mach 1 while being piloted by USAF Maj. John M. Davis, French air ministry spokesman. It is the first French craft to achieve supersonic speed.

British Overseas Airways Corp. has announced intention to purchase five lighter versions of turboprop Bristol Britannas, subject to government approval. Delivery is due in 1958-59.

Canadian Pacific Airlines has been licensed by Canadian government to operate passenger, mail and freight service from Vancouver to Mexico, El Paso, San Paulo and Rio de Janeiro. The service would mark first Canadian route into Central and South America.

New engine fusion, costing \$15 million, has been applied by Canadian Pratt & Whitney Aircraft, Ltd., at Ingersoll, Ontario. Quebec Production has started an R340 Whop engine for Canadian built, North American Aviation-designed Alouett II helicopters.



C.A.V.U.

CONST. AREA FANLEY GUNITE

Large well placed windows give the Chase Assault Transport pilot unobstructed vision as he comes in for a forward area landing.

Delivery of heavy ordnance, transport of personnel and evacuation of wounded from advanced combat zones, without benefit of airstrip or prepared landing field, is routine for the rugged Chase C 123.

Designed and developed specifically to withstand the grueling punishment of combat zone missions, the Chase Assault Transport stands unchallenged in this field.



AVIATION CALENDAR

Nov. 6-7—National fuels and lubricants meeting, Society of Automotive Engineers, The Hyatt, Tulsa, Okla.

Nov. 7—SME symposium on economics on cost, Western Union Auditorium, New York.

Nov. 8—Annual Maintenance Tool Engineering conference, University of Illinois, Urbana, Ill.

Nov. 10-13—English annual convention of The Magnesian Ass'n., Hotel Ritzman, New York.

Nov. 11—Hooper distributors' annual meeting, Lack Hays, Pa.

Nov. 13-25—Acoustical Society of America symposium on aircraft noise, San Diego Club. (For details, write ASA, 57 E. 14 St., New York 22.)

Nov. 17-20—National Aviation Trade Association convention, Hollywood Research Hotel, Los Angeles.

Nov. 18-23—Fourth Annual Safety Seminar sponsored by Flight Safety Foundation, Houston, Texas.

Nov. 20-22—5th Annual meeting of ASME, Hotel Statler and McQueen, New York N. Y.

Dec. 3—Symposium on lightweight heavy weapons and extensions for aircraft, SAE, ASME, IAS and ASME, Hotel Statler, New York.

Dec. 2-5—Aviation Distributors and Manufacturers Assn. trade annual meeting, The Knickerbocker, Miami Beach.

Dec. 3-6—Security for Experimental Stress Analysis, annual meeting, Hotel McQueen, New York.

Dec. 17—Annual Wright Son dinner, 7:30 p.m., Statler Hotel, Washington, D. C. Wright Son dinner to be presented by IAS 3 p.m., U. S. Chamber of Commerce auditorium.

Jan. 12-16—Annual meeting and engineering display of Society of Automotive Engineers, Sheraton Cadillac Hotel, Detroit.

Jan. 14-15—AEE TRENDS conference on High Frequency Measurements, Statler Hotel, Washington, D. C.

Jan. 16-21—Plant Maintenance Conference, Public Auditorium, Cleveland, O.

Jan. 19-21—Winter general meeting of the American Institute of Electrical Engineers, Hotel Statler, New York, N. Y.

PICTURE CREDITS

5, 10—Wally World. 11—Fitzpatrick Aircraft. 12—NACA. 13—Lockheed. 14, 15, 17—Boeing Aircraft Co. 16—NACA.

ALOFT—Great Hughes XH-17 on its first official flight hovers about 40 ft. above the ground at Colver City, Calif. It is powered by two modified General Electric J35 turbojet engines. Air is transferred from the engines through the rotor blades to burners at the blade tips.



Hughes XH-17 Cargo Copter Goes Aloft



PERSONALITIES—Closely connected with development of the XH-17 was Jack to right, Ben Hypper, Hughes Aircraft Aeromedical division director, Howard Hughes Clyde Jones, chief of aeromedical engineering, Western World, meeting Col. Col. E. Jackson, USAF Air Research & Development Command Headquarters, Baltimore, Md., and Phil Cole I. Mason.

GARGANTUA—Two tall 600 cubic footers beneath the XH-17 emphasize the huge craft's dimensions. It stands more than 10 ft. high, its main blades have a diameter of more than 115 ft. The craft is designed to handle large, heavy equipment such as bridge sections, artillery and vehicles for delivery against surface landings such as rivers and mountains.



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Excellent tensile strength and adhesive properties of Polyken Tape No. 225 attach tape to hull and end up on bottom in place before seal can be started with acrylic type bonding by competent.



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WHO'S WHERE

In the Front Office

Harold K. Gray has been designated vice president in charge of The American World Airways' Atlantic division. His former post as vice president Pacific-Alaska division is being filled by vice president Clarence M. Young. Meanwhile Mr. Young has been elected a PAN vice president.

Donald G. Rouse has been named assistant to the president of Rock Airways with offices in Burbank, Calif. He previously was station manager of the carrier's New York terminal.

J. W. Miller, assistant general manager of Midcontinent Airlines, again is in charge with himself. He moved to Dallas Tex. and assumed his duties as vice president of the combined company. He is also a director.

Angel Martin Perez has been named executive vice president of the new Madrid, partially owned Aerovias de Mexico y Central Americas, which reportedly will open routes to Spain, France, Holland and England. Named to the board Manuel Palacios, Eduardo Argente and Leon Garcia Lirio.

Rear Adm. William C. Clausen, USN (Ret.), has joined Bendix Radio division of Bendix Aviation Corp., Burlington, N.J., as staff consultant in the division general manager and Bendix vice president. Clausen served as director of aviation radio for Collins Radio Co. The retired from the Navy in 1946.

Emory B. Kerdin has been designated assistant to the president of Helicopters, Inc. and its subsidiary Loony Construction Co., Inc., N.Y.

Harvey B. Mowbray has joined Shilman Rubber Co., Colton City, Calif., as vice president production. He formerly held a similar post with Los Angeles Standard Rubber Co.

Donald L. Hare has been elected president and a member of the board of American Electronics Mfg., Inc., Los Angeles, a new firm handling precision computer components and instruments.

Harold Galloway, Jr., has been designated vice president in charge of the Air Corps division of Knight Airlines. Galloway has been with Pan American World Airways and Puerto Rico Airways.

Changes

Earl Ross, formerly an executive with Consolidated Value Aircraft Corp., has joined Mic Corp. to advise its related companies in production problems. He will assume available for purchase marketable Air Carrier.

John H. Strick has been named superintending aircraft wheel and manufacturing for B. T. Goodrich with offices at Troy, Ohio.

George T. Keller has been appointed sales manager of the Engine division, Buick, as a representative of Pacific American Corp.'s Sales division. Other changes: Ray Buckman to sales manager, Products dept. and Ernest L. Black, sales manager, Motor & Engine division.

INDUSTRY OBSERVER

►Enthusiasm for the new Kollsman Converter (KCo-2) by-pass engine—filling the gap between turbojet and turboprop—is tempered in some quarters by knowledge of the special problems involved. One example: The engine is planned around a specific requirement, if the requirement changes, a new engine is needed. Installation of ducting, which is present on some engines, makes the conversion of ducting to air with exhaust gases in ducting is complicated and difficult. Another stumbling block is drive mechanisms for the fan, used to boost energy of by-passed air stream.

►Worries for a serious reconsideration of the turbine configuration in turbine lighter reciprocating. Layouts will feature extremely low speed ratios, success gaps between wings and large stingers. Design efforts high strength and stiffness with complete wing area and possible fire impingement. Another application would be a compact aircraft in which the turbine wing served as lifting surfaces and contained the swept engine between them.

►Top engine engineers from United American, Eastern, and Pan American visited General Electric's Lockheed (L40) aircraft gas turbine plant last week for a three-day technical discussion on new and future GE gas turbines which could find use in jet transports. Another purpose would be to discuss the authority to permit GE to disclose recent classified jet designs.

►First production North American Navy FJ-2 carrier-fighter version of the Air Force Sabre is due to fly at NAA's Columbus, Ohio, plant, claim the Air Force, and the Navy, starting about the first of the year. Two prototype FJ-2 which passed carrier tests did not have wing-folding provisions, but those are included in production version.

►Airbus carrier testing on propeller governors equipped with high-pressure hydraulic by-pass "safety valves" in present involvement reversal should be ready to start in November, according to CAA, Minneapolis. Air Line Pilots Assn. has asked CAA to permit wiring for the new changed mechanism to be included in testing that is now taking place in propeller exhaust in another preventive measure.

►Competition aircraft owners poll shows the average business plane owner wants a plane that will carry 15 passengers, cruise at 255 mph., and have pressurization: twelve gals and no less than 72 in. headroom.

►North American's new F-100 version of the Sabre, while most of the new planes being purchased by USAF, was a company-sponsored project, will be the delivery to GE an obvious requirement, rather than an airplane built to meet present military specifications.

►Air Defense Command has notified its Ground Observer Corps to be on the lookout for 840 military balloons in the air with altitudes between 9 ft. up to 71 ft. These balloons are being released for weather observation at various points across the continent. Many are being released at three points along the Pacific Coast to drift eastward across the ocean to 80,000 to 100,000 ft. altitudes.

►Navy observers have been furnished improved with the curly tail flights of the Douglas F4D Skyray despite the fact that the delousing interceptor has been flying with an Allison J35 turbojet instead of the more powerful Westinghouse J46 originally scheduled for prototype and production versions.

►Tail pipes on the jet-powered Hughes KH-47 giant helicopter (picture) page 9—emits from a mechanical drive all the two jet engines and has no gas ducting return like the main rotor. Actually, it is a shrouded tail rotor which drives for a Sikorsky H-35, which was adaptable to the main rotor. Higher clearance because of the absence of torque in the gas drive main rotor.

NATO Air Power—A Staff Report

The air forces of the North Atlantic Treaty Organization are like a skeleton that has recently added some flesh to its bones but still lacks sufficient muscle to wrestle with a formidable foe.

The buildup of NATO air power has accelerated during the past year but it will fall short of its estimated goal of 4,000 tactical aircraft in operation by the end of this year. To measure the scale of the NATO contributions it is interesting to recall Air Force Secretary Thomas P. Switzer's recent disclosure that Russia has provided an Chinese and North Korean ally some 4,500 aircraft, including more than 2,000 jet fighters, and is currently moving toward acquiring a sizable force of Soviet jet fighters in the Korean war. During the recent 1964 Atlantic NATO maneuvers in Germany the Allied Air Forces in Central Europe flew for the largest of the NATO AF command's new period of its record in operating some 1,200 jet aircraft.

► **The Linerup**—The NATO air power lineup now includes three main commands:

• **The Allied Air Forces Northern Europe** with headquarters in Oslo, Norway, and commanded by Major General Willem Carter. This includes the Norwegian and Danish air forces.

• **The Allied Air Forces Central Europe** commanded by Gen. Lucien Nordeur with headquarters at Fontenay-le-Comte, France. NAFCFE includes U.S., British, German, French, Dutch and Belgian units.

• **All Air Forces Southern Europe** with headquarters at Naples, Italy, commanded by Lt. Gen. David Schlatter. This includes Italian, Greek and Turkish units.

Although all three commands are headed by American officers, they are subordinate to their respective commands to a British, German, an American general and an American admiral. Nowhere does an American command exist. In both the northern and southern commands the bulk of the air power actually available in effort should national interests and is not under the air command.

Working on each of eleven nations into a coordinated effort for the purpose of operating under a single command is a formidable task and nobody in the NATO air forces would deny that intercommunal obstacles must still be surmounted before this goal can be achieved.

Perhaps the basic problem is aircraft. Not until the plans to bolster NATO air forces with a significant force of F-85 Sabres are implemented will there be much chance of successfully controlling the air over Europe. At present there are USAF and Canadian Sabre groups based in England and the first contingent of Sabres for the RAF has already reached England. All of these units are outside the NATO commands. For NATO a Canadian Sabre wing is scheduled to move shortly into French and German bases and at least one USAF fighter wing in Germany is scheduled to replace its F-84s with Sabres.

By the end of the year a noticeable Sabre force could be mustered to do battle with MiG-15s, and this command over Europe and provide protection cover for the other units with which the NATO forces

are now equipped. At day fighters they now have British Vampires and Meteors and American F-84 Thunderjets. For night fighters there are Soviet radar-equipped Meteors and Victors. In the bomber category there are only piston-powered B-26s. For transports they have two wings of C-119s and a wing of C-47s.

► **Standardization**—Similar have been made in standardizing communications and field maintenance for the varied nationalities and equipment the NATO air forces are operating. As a result of Operation Double, cross-country techniques were developed so that a squadron of any type fighters can now land at any NATO air base and be serviced and re-equipped to fly another combat mission. A program of standardizing aircraft construction is well along to provide 8,000 lb. runway, but wings developed because actual construction must be done by the country whose the base is located. In France this has caused particular trouble.

Basic supply problem of the NATO air forces are handled by the various countries for their own units. This presents a logistical knot that is admittedly important for sustained combat operations. At present, American air units in Europe are in the process of a long redeployment, withdrawing its bases on the west bank of the Rhine and building up in southeastern France. A major shift in the logistical system to support these forces is also in progress. Instead of the post-war pipeline through German South Sea ports to the air command which was only 90 mi. from Red territory, a new supply line is being built up from French Atlantic ports. This has required an enormous American investment and will require even heavier expenditures if such basic items as adequate jet fuel supply are provided.

The forces that the various NATO countries have contributed to the post-war effort in Europe represent an effort to build a line against the combination of the Red army and its supporting air power in what is possible. But behind this temporary line lie varying national theories on the kind of air power required to counter Soviet aggression.

At the moment there is a strong feeling in Europe that the danger of a general war between Russia and the Western Powers has receded and there is more than one eagerly anticipated to shield western defense. The American viewpoint is still predicated on our ability to deliver a crushing atomic aerial attack as the price deterrent to Soviet aggression.

► **French Views**—The British soon convinced that as strong of a link with the Western Powers must be prepared to fight a "cold peace." As a result, the British are reorganizing their air effort to provide more air transport to enable them to handle their emergency resources to the world's trouble spots and more tactical air power to support their ground forces wherever they may have to fight in new Korea.

The French are bent on building up a military unit to meet their closest trouble, complete with land forces, submarines, a full repertoire of naval and land based aircraft and jet transports. Concerning the French this is an entirely unnecessary and economically impossible is one of the toughest problems faced by NATO planners.

AVIATION WEEK

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F-86 SABRES SWEEP SKIES over Korea, keeping way eyes open for MiG-15s that might make some Yule to attack UN fighter bombers.

Combat Report From Korea

Sabres Still Rule Skies Over MiG Alley

- Scoreboard shows F-86s destroy Russian fighters at 15-1 rate; ratio for all UN planes almost 11-1.
- But basic concept still is air superiority, isolation of battlefield by interdiction, plus close support.

By R. P. (Pepper) Martin
(McGraw-Hill World News)

Tokyo—One of the longest sustained air battles in history is still being fought in the narrow northwest corner of Korea, between MiG Alley. In a sense, it is a battle between a modernized Goliath and an overgrown David. The Red Goliath has superior strength and possibly a better advanced weapons, but victory has gone to his sligher, more duffed opponent.

The Reds are using their fast-burner pilots, but they are not, and third-handers are gradually catching up to them as well. They are mounting a greater number of sorties a day than at any time during the war. The pilots are more aggressive and teamwork between individuals and between flights has greatly improved.

► **Sabes Goliath**—The F-86 Sabre, which bears the brunt of the air fighting, has been improved, but so has the MiG-15. There has been a considerable increase in the thrust of the engine's jet engines. In combat at high altitudes, the MiG-15 is pulling away at a much faster rate. Some pilots

thought the Reds were using after-burners, but that is now generally discounted, since there is no photographic evidence to support the belief.

The Sabres have a "podjet" but it is not yet in wide use, and it does not seem to have played a major role in air combat.

The Reds, apparently dissatisfied

with their chances, have been experimenting with man-to-man gun combat. Most Americans are waiting eagerly for air-to-air duels as the answer to their gunnery problems.

The Reds are fighting differently, where the odds favor them. They are probably getting three to four times as many jet interceptions as the Allied Nations into any single fight. They have excellent ground radar. Fighting close to home, they have a longer period of intense combat. They have a no-fly zone over the Yalu River, to which they are free if land-based. And a damaged Red aircraft has a greater chance of landing safely at its home base only a few minutes from the battle zone.

► **Scoreboard**—Despite their advantages, during the three-month period ending Sept. 10, 107 Russian-type MiG-15s were destroyed in air combat. The U.S. Air Force lost seven F-86 Sabres and three F-84 Thunderjets in the same battles. The F-86 superiority in combat is better than 15 to 1, while the odds of all UN planes in air combat are almost 11 to 1. Figures for October are not yet available, but the ratio may be even higher from the three-month ratio.

When, then, is the nature of David's strength, to the extent of Goliath's comparative weakness?

The major factors accounting for UN superiority are present air pilot efficiency, training and gunnery, rather

New AW Writer

Robert P. (Pepper) Martin is serving as Associate Writer Tokyo correspondent while A. W. (Pepper) chief of McGraw-Hill World News Tokyo bureau, is on leave of absence. A war correspondent in World War II in the Chinese theater, Martin has been in Japan for the Columbia Broadcasting System there and a half year. He returned to Tokyo recently after completion of a Nieman Fellowship at Harvard. Joseph is studying at Princeton under a research fellowship awarded by the Council for Foreign Relations.



SCOP INTAKE'S dogleg outlet took No. 1 North American YF-93A while



FRESH INTAKES are fitted on No. 2 plane used by NACA for experiments.

Four Planes From Two YF-93As

Fast jets supplement windtunnels in NACA studies on inlets for missiles and high-speed aircraft.

A pair of jet guns identical, two-to-high speed, sweeping jet airplanes—also opening up new avenues in flight-testing for the Ames Laboratory of National Advisory Committee for Aeronautics at Milledge Park, Calif. North American YF-93As, were transferred to NACA a few months ago by the Air Force. Their unique value as tools for flight testing lies in the fact that they are identical planes except for two important differences.

No. 1 plane has scoop air inlets on the sides of the nose, while No. 2 plane has flush inlets in the same location, the first is fitted with an afterburner, the second is not.

With the simple expedient of transferring the nose section of one plane with the tail section of the other, NACA flight test staff can get a total of four different airplanes to use in its air-flow flight-testing program, with virtually identical characteristics except for the planned variations.

Most of the Ames-NACA research has been studying various types of inlets for the thousands of cubic feet of air which the jets inhale. Prior to the development of jets the researchers were studying air intakes for use in piston engines.

As of now, they have most of the answers for an inlet for high subsonic

and perhaps transonic flight, but their research is continuing on inlets for higher speed flight of both aircraft and missiles.

Most of their research is that field that has been in uncharted flight in high speed windtunnels.

Now the YF-93As make it possible for them to correlate and extend their windtunnel data with flight tests, providing supplementary information they could not get in the existing windtunnels in full-scale conditions.

These two airplanes are powered with Pratt & Whitney J46 engines, with 6,530 lb thrust rating dry, and capable of about 9,000 lb thrust with after burner. Although a change in Air Force requirements diminished quantity as data for the F-93, there are two airplanes in service today that are in fact, or they are equipped with such a power full engine.

Airflow Problems—NACA also has a test North American F-86 jet plane, from which the YF-93A design was largely developed, which gives the research organization a nose-inlet airplane; with somewhat smaller characteristics for another basic of flight comparison with the other inlet types.

In simple terms, the problem of high speed inflow for jet aircraft involves the ramming of three separate streams of air

• The outside air through which the plane is moving.

• The main stream of air which enters the engine at the main intakes and passes through the engine to be exhausted at the tailpipe.

• A supplementary stream of cooling air which enters through smaller inlets farther back on the fuselage and also exits through the tailpipe.

The basic problem is to make the air flow into the inlets and out of the exit as smoothly as possible. Complications are numerous.

Aspects act differently at different speeds. Flow characteristics differ with and without afterburners. And the way the exhaust gases with the outside air at the exit, where the actual thrust is delivered, is one problem that is especially vexatious.

• **Secondary**—In fact, NACA has found that the biggest difference in high subsonic speed flight performance comes as a result of varying exit conditions rather than inlet conditions.

In some cases NACA has found that any one of four inlet designs will do an adequate job for the subsonic jet airplane, if proper attention is given to design detail. These include the three types of inlets on the North American plane, plus the small nose root inlets found on several other U.S. jets.

However, as the speeds go higher more the treatment given and into the supersonic regime, the design of the inlet becomes an extremely critical factor in the overall performance of aircraft or missile.

One interesting feature of the flush inlets on the No. 2 YF-93A is a blind off for boundary layer air designed to smooth out the main flow through the inlet by separating out the turbulent air next to the fuselage skin.

NACA scientists say that this is an "overs" which these inlets had but which would not be necessary for flush inlets, but that possibly North American designers included it as an added measure of smooth flow.

• **Pratt & Whitney**—In last one other U.S. experimental airplane has been fitted with flush inlets a Republic XF-84 which needed the space occupied by the usual F-93 nose inlet for radar.

The Republic flush inlet plane was developed as a private venture and made several flights, but did not win an Air Force production order. Air Show attendance saw its lastings at Detroit last September in part at Air National Guard's three visitors exhibit. Apparently, Republic decided the flush inlets were less satisfactory than sweptback intakes, at least the new RF-84F is sweeping plane with a solid nose, one for two sweptback inlets instead of flush ones.

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AFTERBURNER on the Lockheed F-94B lights speedily gives extra jet power when needed for increased interception, fighting low...

Engineers Meet Afterburner Challenge

- There's much more to getting that extra thrust out of the engine than just extending the tailpipe.
- Solar Aircraft Corp. engineer tells of techniques used to beat high-heat and combustion problems.

The afterburner has been in the jet engine picture for a good part of the turbo powerplant's history. A lot of data on the afterburner's thrust-adding potentialities, how it works and its general upkeep has been reflected.

But there has been very little detail on specific design and construction.

Not just a Pipe-Contrary to a well-belief, even an aviation engine, the after burner is not just a simple pipe. Essentially, it is a special engine designed to the least wasteful, breathing its exhaust and burning additional fuel for thrust boost required during takeoff, climb or combat maneuvers. As a special engine, it is treated full of design intricacies. And it still allows a heavy loaded engineering challenge.

Early work on combustion and jet flow characteristics of the afterburner was reported in *AVIATION WEEK* Jan. 24, p. 21. Now, more steps have been taken off the jet runway, for a close look inside. At the recent National Aeronautics Meeting in Los Angeles, of the Society of Automotive Engineers, a detailed analysis of afterburner analysis and problems was unveiled by Ralph Kren, design engineer in Solar Aircraft Co.'s Development Engineering Division.

Heat Problems—The afterburner involves special design considerations. To get a substantial amount of added thrust, the gas in the jet's tailpipe must reach more than 3,000° after combustion. Thus it must flow 500° above the melting point of the best alloys.

Then, the tailpipe and components introduce complex problems in high-temperature engineering for universality and reliability.

Detail design needs a special look to assure that all hot surfaces be cooled—usually the high heat gases would quickly melt an uncooled surface.

Another big task is to get smooth, efficient combustion of afterburner fuel. With gas velocities of 300-600 mph, combustion instability produces a force that might cause severe troubles.

When afterburners were first hooked to jet engines in planes, it was soon shown that major improvements were desirable to improve afterburner installed performance. Kren points out. Refinement of basic afterburner design and improvement of performance and efficiency were indicated.

Nozzle Studies—One of the major problems concerned the variable-area jet nozzle. Aircraft with afterburners had to have a smooth variation of pressure from the inlet to the full expanded thrust, instead of the surge experienced with the early two-position nozzle

afterburners. Also, exhaust engine design would give better fuel efficiency by using the afterburner nozzle for no burning operation. These factors pointed up the desirability of the fully variable-area jet nozzle.

Newcomer afterburner two cases in for their share of study, because experience showed that no part of the afterburner is so small as to be considered unimportant.

Kren analyzes afterburner design in considering the installation at consisting of four major components—diffuser, burner, control and variable nozzle. This first to end approach gives a good sequence picture of the how and why of afterburner working.

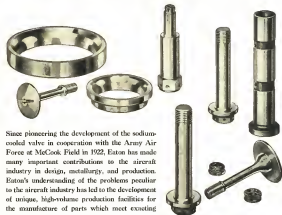
Diffuser—Gases leaving the jet engine's turbine wheel are led through the shortest distance to the jet nozzle. This is done with a streamlined intake directed to the turbine fan.

Leaving the turbine wheel, the gases usually have not come quite as much too high for afterburning. To cut this exit speed to proper value, a diffuser section is required before the after burner combustion. Guarded position Kren says, is to diffuse the gas to a speed of 400-500 feet per second at the combustion section. The diffuser results in made natural with the turbine to not down an afterburner tailpipe length.

A cross section, typical of today's tail nozzles (Fig. 1) discloses that it has an outer shell with flanges for attachment to the turbine frame and balance of the afterburner, an inner cone, support struts and fairings. There also may be a ducting system for introducing the

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Fig. 1. Construction of typical turbine



Fig. 2. An integral turbine design



Fig. 3. A specially modified steel turbine

configuration. Loads taken by the case usually are light and primarily are due to the differential gas pressure across the case. The resultant case gas pressure forces the turbine to rotate.

Reaction: In the afterburner, case burner sections, fuel is injected into the specially high-gas-pressure constant exhaust gas flow the turbine, and the turbine turned to raise the energy level of the gas stream. This burner section usually consists of two super compressor-fuel injection system and flame holder. Frequently the burner section is incorporated within either the diffuser or the nozzle section to save weight and avoid complexity. Kerosene is used. For maintenance and replacement ease, it is more convenient to have the burner a separate section.

Fuel injection generally is through a number of spray bars in the gas

stream, which allows and pulls the spray of oxygen, air, and combustion and exhaust.

The entire case is mounted off center, but as in the motor shell, it must be on axis, both to the shell. The case circular movement is not rigid, allowing a floating action to take care of differential expansion. The draft-like bearings carrying the casings serve to center the gas flow.

Integral Midship: Another type of turbine, an integral design (Fig. 3), has offset main of the rotating and production difficulties found in other case types. Kerosene is used.

When the case is integral with the turbine, the latter becomes structural member instead of bearings permitting the elimination of the structural case.

Rotation of case, to shell usually is by means of gas injected from the jet into, with liquid clearance, to fill their end expansion of the parts.

The case shell mechanism is in two parts: a positive, very heavy, built in no-bearing and shrouding, turbine. Actual gas loading due to pressure difference within the turbine, generally is not very high, indicating that a relatively light shrouding member is required. But this is not true, experience has shown, because fuel and hot oil (due to gas pressure) is a long turbine at case and shell fractures. Great study has shown that these, probably are caused by wear, bending and twisting produced on the shell by the rotating large and heavy case subjected to variations in pressure and vibrations during operation.

With shrouding this condition is much more severe, leading to bearing material that is very short time, kerosene.

Friction Fit: In fact, this difficulty is attenuated and designed in with the case from the shell with high temperature, resistant contact parts. The case to shell connection uses this, and gas, but the resultant gas pressure on the shell and gas. This method has proved very successful, eliminating the tendency of the shell to fracture-perforate in afterburner turbine—Kerosene.

Engines are built assemblies made of 321 stainless steel.

But the high-pressure design standard, the longitudinal or beam stresses in the outer shell, especially as engine design, Kerosene is used. In this case, the use of the longitudinal stress—stresses in G-bearing requirements of the engine—indicates a higher strength material than 321 (normal) W and N 155 are typical materials used in the shell in this case.

Inner case materials, the are 321 stainless-steel, except where burner design carries into the turbine

strain gage



Strain gage equipment is supplied with the Model SG-3 welded metal case, wire strain gage. It may be used repeatedly without disconnection and is ready for immediate operation upon engaging its lead edge contacts to the specimen. Two active bridge arms are employed. The gage factor of approximately 2.0 permits operation with conventional existing resistance strain gage equipment. Two sets of lead edges provide strain measurement ranges of ± 0.003 and ± 0.012 in/in.



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stems, with performance adjusted to give a particularly fast pattern. Used singly in the spray bars it is a tubular manifold. When dual injection system are used, more than one manifold is needed, the manifolds being either external or internal.

With external manifold (Fig. 3) the gas flow area is left clear. But because of the many connections between spray bars and manifolds, there is a possibility of joint leakage—a serious fuel loss.

With internal manifolding, there is only one, or possibly two, connections for fuel to the afterburner, thus cutting down on leakage risk. Also joints are welded, and if cracks do develop, fuel leakage is into the gas stream, which offers no hazard.

Internal Arrangements—The internally manifolded fuel injector (Fig. 4) brings in the problem of differential expansion within the seat, caused by the non-homogeneity of temperature profile across the gas stream and these differentials within the manifold resulting from fuel flow.

In smaller injection rings, Kress points out, this condition is neglected, materials strong enough to withstand the thermal stresses being used. But in larger rings, this design leads to short life, and some provision such as a slip-grooved collar must be used to take care of thermal expansion in the manifold itself.

Other designs incorporate the fuel injectors in the tolerance so the afterburner has no overall ductor length. Here the manifolding is made the turbine, using the leakage problem (Fig. 5). Fuel leakage in this system isn't a danger for the engine compartment but now has in the turbine, Kress says, not been and before the cone.

Fuel injector size is selected to pass the fuel at optimum initial velocity,



Fig. 4. Internally manifolded spray bar



Fig. 5. Three fuel injection are tolerance-managed to keep afterburner length down



Fig. 6. Diagram of flangeless installation mounted downstream of fuel injector

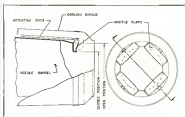


Fig. 7. Detailed drawing of a four-lap variable mode for afterburner

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—T. C. Kane, Chief Engineer,
Commercial Shearing & Stamping Co.

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or that tenders, in fact, tend to be maintained, not put into excessive pressure.

Lighter cross sections are usually a plus, but coolant gas flow turbulence sometimes causes detachment of flame to the injector, bringing across one sudden disturbance, leading to failure of the injector and other important parts. Automatically introduced injectors have been devised which almost entirely eliminate this problem. The action must withstand internal fuel pressure without permanent deformation, because pressures may reach 100,000 psi.

Loads imposed on fuel injectors result chiefly from gas speed and are relatively light. Injector material needs is AISI 121 corrosion resistant steel. **Flameholders**—The unit in Fig. 4 resists disturbance of the fuel injector at a distance to give proper fuel dispersion, maintains a stable flame front across the turbine. It is a turbine, producing device which compresses gases from gas speed. Its flame rate is separated in the design at that time, flames do not come substantially there from the turbulence is not quenching or that third gas is not seen with reduced with turbulence.

The flameholder is a, suggested from the difference size but also zones must be made for differential expansion—flameholder resists very high temperatures while the gas resists relatively cool. Support a nozzle through external, suitable pins, through threaded bases, to a knee clip fit on mounting on the flameholder.

Stress between flameholder and outer shell must be mechanical. Kerosene, in order turbulence which tends to propagate flame, causing that tends to help the flame. Gas turbulence or gas turbulence in the gas stream must be checked to eliminate hot streaks. These define the turbine, during the turbine, during the turbine.

To eliminate hot streaks from turbine flameholders have been incorporated into internally manifolded fuel injectors, also off the main case.

In each case, the parts must be manufactured to take the lowest loads brought by violent combustion forces in the atmosphere. Flameholders are made of N 115 or other high heat alloy, rather than being being needed. They are usually fabricated surfaces but not with heat treat.

Nozzle—A jet engine without afterburner, nozzle diameter is chosen to give best operating efficiency over a wide range of flight conditions. In an afterburning jet plant, the diameter range required of the nozzle is so great that a variable nozzle is used.

With afterburner nozzle size of the turbine type. Disk, one flameholder nozzle size not suitable, for

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Fig. 8 Shows a full-scale working model of a flap variable moule.

the maximum possible thrust. In this type, two stainless steel claustrals are hinged at the end of the tongue, so arranged that they retract the flap in the closed position, opposing exit of the air from the gas stream in the open position, thus affording the moule area change required for afterburning operation.

Keen interest in adoption of the claustrals was to fully variable operation. The separate fuel very high strength freon and superalloy making, because of distorted parts, resulted in poor performance and rapid deterioration.

An investigation was begun on the design of a fully variable flap-type moule. Preliminary studies showed good possibilities for a design that would be relatively free from distortion and also adaptable to production processes.

As a first study, a moule was designed that incorporated four large flaps, which were so made that as each were raised and lowered were adequate at turning, temperature. Keen inquiry studies showed that the loading would not be excessive, although too high for the simple air cylinder actuator then used.

Fig. 7—Fig. 7 shows the flap-type moule. The flap is hinged at the end of the tongue, so arranged that they retract the flap in the closed position. The flap project into the gas stream at right angles, to a circular shape when open. Flap operation is by pistons connected to mechanism, which is mounted within the body of the moule. The long inlet hose, as low as possible, the gas stream end of the flap is generally rounded.

An actual full scale working model of the moule design (Fig. 8) was constructed of these parts, and spot welded together. Nozzle barrel included ing hinges, are 316 and 316 are N-455 spotwelded assemblies. Flap is the piston type, each half section welded in place. Because of the high temperature and the gas stream, the flap is parallel to the gas

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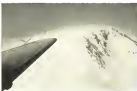
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Figure 9. Visible bulb mode, in closed position.

shown, it was felt that no seals would be required.

Tested more for many afterburning and no-burner tests (shown) to great degree of determination. Nozzle losses were low, with slightly better efficiency in the latter open position.

Major disadvantage of this design, Kren reports, was the relatively large clearance diameter required in the inlet pipe at the jet engine. Because complete base drag is a function of the base vacuum area between the jet stream and the surface upstream, it was felt that further seal, should aim to make smaller most desirably efficient.

Other Approaches—A nozzle in which the subject area is entered by a bulb moved in or out has also been under consideration (Fig. 8). This has shown some success in a non-afterburning engine, but when used in the relatively hot stream of the afterburner, the action problem—together with the structural loads induced in the bulb—presented an extremely difficult problem, Kren recalls.

Additional studies led to the requirement for a larger number of individual flaps which would give a more circular inlet and be placed in all positions. This would allow a more free shape that could lead to high nozzle coefficients.

Kren reports that numerous preliminary nozzle design have been developed over several years, none having succeeded many times on the test stand in both afterburning and non-afterburning conditions. They have proved, he avers, that the multiple nozzle can be extremely rugged, and it can be manufacturing crash.

Adjustment—Fully afterburner modes of the Cleveland type, having two positions (Bulb open or fully closed), allowed an actuator adjustable only to these two positions. The nozzle was automatically balanced, so that no ad-



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INSTRUMENTS
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actuators (which are used to control the fuel) (which) against the closed position and. Compressor air usually was enough for the job.

But the variable-thrust afterburners, which at full variable area nozzle, made necessary an actuator reliably adjustable from fully open to fully closed and also capable of locking at any point in the range. Kerosene ports out for the fully variable actuator, greater power because a pneumatic response, because as like the balanced chamber the variable flap-type nozzle is unbalanced and the actuator must overcome both friction and internal gas pressure acting on the nozzle segments.

► **Electrical Actuators**—On a jet engine, actuators are limited to four practical sources of energy—electricity, hydraulic pressure, lubricating oil pressure and pneumatic pressure. Each has its pros and cons, none is ideal.

Kerosene is an electrical actuator is a natural choice for an afterburner—it has been made for many years in all shapes, sizes and for a variety of operating conditions. electrical energy is readily available in a plane is easy to control, and actuators are easily designed into electric control systems.

But because the actuator must be placed near the variable nozzle and the high heat involved, the electrical unit needs complicated cooling means. Also, power requirements of flap-type nozzles are about 7 hp, and up and this probably is the most serious drawback of the electrical actuator, because ratio of the power are not known to be "high-weights," he says. The electrical actuator has been tested more or less to the hydraulic-type flap-type nozzle.

► **Hydraulic Actuator**—The hydraulic actuator, high pressure installations can provide high forces with small lightweight actuators. However, the hand detents particular rate in growth and packing, and all have and require subject to leakage must be kept close of the hot afterburner surfaces.

High ambient operating temperature of the jet engine is a serious problem in the operation of many types of control valves used with the hydraulic actuator. Kerosene says. The hydraulic fluid could boil or decompose under high heat. But with the new high-temperature hydraulic fluid, this problem could be alleviated, he claims.

All jet engines have pneumatic lubricating systems, which could be used as sources of energy for afterburner nozzle actuators, but limited forces would be produced because of the actuator's low recovery, he adds.

► **Pneumatic**—It is also can provide reliable high pressure air from the compressor section, for actuator power. Kerosene holds that quite a bit of an area be used before serious loss of power is noted in the engine. He considers

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Brazil Division, Rio de Janeiro, Brazil

compressor discharge as attractive as a power source. It is always available while the engine is running, exhaust air may be dumped almost anywhere without danger, a machine works as well as it receives or returns heat as readily as it is utilized to its great extent by high pressure, design of packings and seals could be relatively simple. Leakage introduces no safety problems, and because air is compressible, it is a source of potential energy.

Disadvantages, he asserts, are serious, but not too concerning. Available compressor discharge pressure in most cases is relatively low in comparison to a standard hydraulic system. This pro-

cess also is a function of altitude and speed. Also, because air compresses, it acts like a spring—when used in actuating cylinders, it gives the piston bouncing action. Thus, positioning of the piston is very difficult.

► **Actuator Answer**—Selle has designed and built a practical actuator that is valve stabilization of an air cylinder to make it fully pneumatic and act like a hydraulic cylinder, Koss reports. It incorporates an integrally sealed and self-contained hydraulic system.

This actuator is of the screw type and is designed to operate at an ambient temperature of 600°F. Hydraulic system, including fluid and seals, is composed

of materials capable of withstanding this temperature.

Pneumatic action is fabricated of stainless steel and aluminum with special piston rings designed for leakage-tight operation. Aircraft quickly with as low as possible weight output into as other features.

—Living Stone

Saucer Shape Seen Best for Spaceship

There is a logical and acceptable system for a flying saucer, says Dr. W. F. Hilton, well-known British aerodynamicist, because such a shape appears to offer the best solution to the problem of landing within the atmosphere of a planet.

What you need for best deceleration, assuming that braking rockets are not used, is a vehicle with maximum drag per unit area, he says. Spheres, which show such drag, are out because you also need lift. Then a disk, rotated for stability, and thick at the center to take payload, seems to be promising for certain classes of spaceship.

► **The Problem**—Dr. Hilton made these points at a recent meeting of the Midlands branch of the Royal Aeronautical Society, where he considered the aerodynamic problems of landing and taking off.

Since most of any interplanetary flight would be spent in space—where vehicle shape is unimportant—it would be the disk that spent in the atmosphere that would dictate the design of the craft.

Takeoff is a negligible factor in design of spacecraft, Hilton feels. Take off would be vertical and from the highest point available to maximize drag losses. Acceleration would be low, and no serious aerodynamic problems would arise, he says.

But on the return journey the arrival velocity would be the most at that of escape—7 miles per second, roughly—and the vehicle would make contact with the upper layers of the atmosphere at a Mach number of about 35. As one gets to the neighborhood of the problem, Hilton states that a steel bullet will melt at a Mach number of 5.5 in the absence of thermal radiation.

Thus a returning spaceship will have to reduce speed from Mach 35 to Mach 5 before sustained flight in the atmosphere is possible.

► **Path Selection**—To slow down most efficiently, the spaceship must make grazing contact with the atmosphere, skimming as much heat as either the ship or occupants can take, and then climb out to reenter heat into space. Contact as such an orbit—made by that of a hot stone slipped across the surface of water—will be made on the disk side of the planet to minimize the



The Republic Thunderbolt has an amazing record of withstanding heavy battle damage and still returning its pilot safely to base.

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available inoperative differences. Both a "slipping" flight path may actually proposed by Dr. Peter Sauer, in his wartime studies of a hydrogen bomber like the Geyser. Sauer's intent was to get a lot more by using of this technique, rather than to transfer heat efficiently.

Hilton proposes that the metal coils be shut-in, or two strands in a electrical coil of several turns—and that the aluminum foil should be directed towards the plane in order to increase the heat in the atmosphere. **■ What?** Skipp—Contrary to actual sword design, the spinning will heat to three thousand deg per unit of exposed area. Of all geometric solids, the sphere produces the highest drag per unit of surface area, but has no lift. A circle in space shape at a high angle of incidence would be most satisfactory, and the drag produced could be turned off by flying at zero angle of attack.

Solid shapes would be slightly as well, but if a disk were used, it could be rotated for spin stability about a perpendicular axis through its center. The spinning would be quite thick through the center to take the payload, and then at the edges the general concept of a flying saucer.

Placing surface of the saucer would be alternate layers of steel and asbestos. Most of the aerodynamic heating would occur behind the strong inboard shock wave on the lower surface, the upper surface would be little affected. **■ Analysis:** Hilton suggests that an angle of attack between 10 and 30 deg would be most satisfactory. This would produce a lift coefficient of unity and a drag coefficient of 0.27.

All the vehicle's kinetic energy would be converted into heat, and at low angles of attack that heat would appear as the vehicle shock. At high angles, the strong shock went out below would heat the air passing through it, thus much of the kinetic energy would heat as smoke from the propeller.

Dr. Hilton said that hydrogenated water would be needed before it could be used that the hydrogen disk was the perfect answer, but that it looked very promising as a shuttle vehicle between a plane and its satellite. —John Humphrey

P&W Metalsmith Course

A jet engine metalsmith training you gain began in January 1950 at Post & Whitney Aircraft has become first class. For those men have completed the 6,500 hr apprenticeship course to create skilled metalsmiths for special P&W production jobs.

Aircraft group of 72 men are still in training, and a third group of 26 began the program in October.

Castings Seen Best For Missile Casings

An Irbis aircraft is being put on casting for guided missiles.

By Gen William T. Hefley, ANIC assistant for material program division, recently told the American Iron & Steel Institute. Society that casting—the cheapest method of fabrication known—will provide a cheap, disposable shell for this type of weapon. For quantity production, the Air Force will be looking for the least expensive, dependable materials that will do the job, and Hefley said that casting points to

the foundry industry as the place to get 6-8 of the nation's casting companies has a long way with defense needs.

At Fort, thinking on missile components involved, improved techniques in gas, strength, lighter, more heat resistant castings at reduced costs in materials and low materials in major factors in the missile program.

Hefley reported that aluminum castings, capable of taking pressures of 70,000 psi, will be produced soon. Also, a recent experimental project involving Northrop Aircraft, Inc. and Aluminum Company of America, was for casting an F-102 magnesium alloy, 14 ft long wing, with existing technology.

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THRUST & DRAG

The second Space Travel symposium held at the Harvard Museum recently produced a fine technical symposium over which we'd be getting to the moon. The group was divided into a 15-yr time lag group and a considerably larger, unspecified time-lag group. Not making it into the latter side of the moment, T&D would like to point out a few angles to those who feel that it can be done starting now on today's technology with lots of money.

The physical rocket could be built today's not spent with that position. But the supporting infrastructure and logistics and fuel costs never are figured into the overall sum. Test flights might cost a couple of million dollars each, and you wouldn't get more with much on other test in a dozen.

And as for the time in today's knowledge, consider 30 to 40 years ago. If the Wright brothers had been given \$1 billion, their success could have produced a single V-2 rocket. If Russians had been given the billions, the rocket could have developed earlier.

Scientific progress is made by extrapolation with occasional flashes of insight.

solid gases along the line. The general trend of the program curve against time is consistent, not inconsistent. We'll get to the moon—but we can't start now.

Tim Bets By, honorary engineering organization, has again delivered a resolution to adult women engineers to that learned organization. However they will continue to avoid the Woman's Budget to outstanding job in engineering.

Just what Tim Bets By hopes to gain by this is hard to see. I consider such a stand by the society is indefensible. There are many women engineers in the profession today. I've worked with some and been glad of their technical prowess and their decorative value around an otherwise dull office.

The only qualification for membership in a professional organization should be competence. It doesn't matter if the applicant is a beautiful blonde or a blond to happen some day a girl becomes God's best and complete her.

Anybody who wanted to see a model of the Navy's new secret air engine, the Space Sparrow, could have done so—just a short time ago—by looking at the window of Arlington's restaurant on Connecticut Ave. in Washington. A model of one of the Douglas AD series was displayed complete with two new mounted Sparrows. So that there could be no mistake, the birds were lettered with the correct designation. This is society?

William A. Lanes has come a long way from the original Douglas M41 airplane which led off his successful first 20 years ago. And with William announcing the full details of his latest new Douglas DC-6B, the comparison between these large new vehicles and the little M41 is extremely interesting.

The total load capacity of the M41 is 1,775 lb. at gross weight, and, fuel and instruments was less than the weight of the electrical system of the DC-6B (1,783 lb.).

The DC-6B at \$1,850,000 costs about 280 times the M41 price of \$11,000. The 400-hp. engine of the M41 pulled it through the air at a whitening 115 mph., today, the 9,000 hp. available in the DC-6B produces a top speed of 360 mph. The M41 could be handled by one man, and it carried one passenger; the DC-6B needs a crew of five, but carries 66 people.

And the associated paperwork, his informed, too. Space for the M41 took eight inspection pages, but the complete specifications for the DC-6B require a 176-page volume. —BMA

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The Re-evaluation of Load Ratings for Airframe Control Bearings



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Selection of control bearings is made

solely on the basis of the bearing's static "non-fretted" (KNF) value only — ignoring completely such factors as normal or combined loads, differences in applications, and cycles of excitation. This method therefore does not provide an accurate rating of individual bearing capacities.

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Titanium: Headache With a Future

- GE notes drawbacks as well as advantages.
- Ductility, cost, quality, are unsolved problems.

Titanium has established itself as every important member of the family of metals in engine applications. But the more other metals with promising general characteristics, getting down to specific engine applications has posed problems.

Data on one of these applications—the use of titanium forgings for jet engine compressor wheels—has been compiled by L. R. Foster, General Electric Co. metallurgical engineer.

Requirements. Foster outlines GE's findings as the light of material requirements for jet engines, with regard to weight, corrosion resistance, and size advantages.

Titanium fulfills these requirements, but, Foster emphasizes, jet metals also must have ductility with strength, an easy quench, after treatment lubrication and eventual low cost. These requirements, Foster says, titanium will fulfill only with the most diligent attention of supply and cost.

Keyed to these requirements, he offers findings on jet engine parts made from titanium forgings. These parts, currently under consideration at GE, include compressor wheels, blades and a portion of the compressor housing, and smaller parts such as pins, couplings and bearing housings.

• **Compressor Wheels.** For these forged components, strength and ductility are needed. The latter is highly important because of rubbing parts.

At the strength levels of commercial alloys such as Ti-15A and RC110B, GE has designed wheels which can save 100 to 200 lb. per engine.

On the ductility side, tests on direct tensile, torsion, Ti-15A titanium disks have shown a broad range of speeds for burning. Foster reports that this has been associated for largely a difference in ductility. Elsewhere, highly stressed part can rupture with local plastic deformation, stresses are locally to the fracture level at bearing points and section change.

• **Compressor Plates.** Compositions of steel ductility, more through variations within an alloy than through surface treatments alloys.



COMPRESSOR WHEELS of titanium are light, but hard to fabricate

Heating and working schedules during forging also affect ductility. As an example, Foster reports that GE's Ti alloys exhibit a grain boundary brittleness after heating above 1650° F. The large grain resulting from high heating appear to be ductile, he says, but the ductility can reach 100% along the grain boundaries.

Excessive a constraint is applied to the beta grains to the boundary as soon as the grain boundaries. Brittleness from heating in the alpha-beta region is the usual case. When variations, probably in composition, have occurred, the potential heating to the alpha-beta range without embrittlement.

Twenty points out that if a large forging induction of the order of 50% decrease in thickness is made while the titanium alloy is cooling down to the alpha-beta range the beta boundaries are broken up and ductility is restored to the finished forging even if forging was begun above 1,500° F.

Many of GE's earlier disk forgings were made with severe restrictions in the number of 1,800°. Most of those, Foster says, showed the intergranular brittleness—more being worst than others. He believes that the beta boundary brittleness is caused by some element not held in the normal composition of the alloy.

• **Quality.** Foster stresses several qualities in titanium products as an essential to successful applications. For disk forgings, ductility must be clean, sound and ductile. Discontinuities such as trapped or trapped inclusions or segregation from incomplete fusion of other additions, he contends, can drastically lower ductility in local regions. GE tests have shown definitely that broken trapped particles actually lower ductility.

Titanium melted under graphite electrodes also has shown typical defects

in ductility. While there are in fact more than with tungsten, Foster says, they represent a hazard which can be tolerated only while the aircraft perfect their consumable electrode techniques.

A disk forging produced with a good surface, well-filled cavities, and adequate properties is still no good, he contends, if there are internal discolorations. Titanium exposure on the part of the surface and even a few inches, he warns, to avoid finished wheels with internal defects.

• **Fabrication.** Practical fabrication of titanium wheels from forgings will depend on machining techniques to start over with difficulties in metal pickup, excessive tool wear, coolant problems, etc.

Foster says that rough machining could be greatly aided by some method for removing the forging surface, by true actual machining, a standard. The one-handled layer below the forging scale, probably causes more good to machinists than any other angle better, he says.

The 100 to 200 lb. that can be used per engine by using titanium alloys in compressor wheels is an attractive advantage. But at present costs of finished titanium parts it probably will not yet to make the substitution, Foster points out. A customer would have to be willing to pay about \$12,000 more for an engine to gain titanium compressor wheels, he says.

• **Cost.** Cost-based avenues of cost reduction are open.

• If sufficient excess quantity per device, substantial price cuts probably will be possible.

• More efficient use of material in keeping through the requirements is another approach.

• Quality control, too, Foster says, "If we can keep fewer parts for internal



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The front undercarriage comprises a light one alloy casting. This new alloy, the REP 6, performed in the

MESSIER assembly, possesses very advanced characteristics. It is to be noted that the dimension control and anti-friction device are contained within the casting.

As MESSIER are responsible for the entire hydraulic equipment of the aircraft, they have taken an overall view of each operational detail to assure maximum efficiency and security with a minimum of weight.

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Here's how the BECKMAN EASE COMPUTER helped simplify F-86 Sabre jet design



To minimize the high costs of designing controls through wind flight tests, North American Aviation now employs control units of the Beckman Ease Computer to perform design while still on the board.

A Typical Problem: To develop an automatic velocity system that would eliminate yaw or side-slipping conditions in gliding the F-86 Sabre jet over a wide range of speeds and at altitudes from sea level to the stratosphere.

How North American solved it: The diagram above shows how North American used control units of the Beckman Ease Computer to quickly solve the problem by flight simulation. A control-equivalent machine was designed by engineers at North American which generated velocity proportional to airspeed and rudder deflection rate by movement of moving shafts and pulleys. These voltages were fed into the computer so that at electrical response was analogous to the output of the F-86 D in flight. Flight conditions—speed and altitude—were varied on the computer by merely turning knobs.

Advances performance confirmed the results as developed by flight simulation.

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COMPRESSOR BLADES of turbine alloy offer weight saving opportunity, at about 3,000 are used in a jet engine.



GEARS still pose fabrication problems.

designs and can obtain forgings with less material to be machined off, we shall have many more forgings for our needs."

Progress with these factors of price, thickness and quality should make it possible to produce forged titanium wheels within economic ranges, he says.

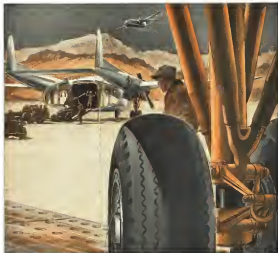
Compressor Casings—Application of titanium forgings for compressor casings opens weight-saving possibilities second only to that of compressor wheels.

Toward the better end of the cost pyramid, cast titanium alloys compare rather badly under operating stresses. Rather than use steel for such a large heavy part, GE is trying to use titanium. Several designs for forming and welding have been suggested, but Pinner says:

"We know of no method to produce double flange welds in the high-strength titanium alloys. Consequently, the welded design will have to stay on the shelf until weldable alloys appear."

Meanwhile, GE is making casings by casting forgings of quadrants from titanium alloy plate. This obviates forging of the outside of the casing has made fabrication more easily perfect.

The casing also must have strength and ductility. Ductility is important because there are many stress concentrations in these longitudinal structural members. Surface quality is important because the outer surface is stretched de-



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ing lagging, and the rotor surface must have sufficient material for machining to accurate dimensions without too much waste. The machining of the casing is a "terrific" job, because any error as well as the price of the material makes the cost of the casing enormous, much too high.

► **Blade Applications**—Use of titanium alloy and titanium blades could be a large advantage in weight saving. Each blade weighs under a few ounces, but considering the total number—nearly 2,000 per engine—weight cut adds up to major pounds.

As a result of changes in direction, size and equipment of turbine manufacturers within the national driving lagging, ductile titanium alloy has not given very little blades. But because of severe stresses imposed during blade assembly and under operating conditions, some ductility is required. GE tests every blade for ductility and is trying to arrive at the acceptance tests which can be met by suppliers and give ductile blades.

Fraser reports that GE has never made a blade blade from the casing, nor before Rockwell C-16. But, he says, hardness is not well correlated with ductility—at the higher hardness of Rockwell C-39 some blades are very brittle, while others have more than adequate ductility.

Careful selection of bar, control of lagging schedule and testing of all units

have proved that assembly ductile titanium blades as part in GE engines.

It appears that there is sufficient ductility in stress levels for static and free blades that the two may be such from difficult titanium alloys, says Fraser. Static blades have an load and will tolerate stresses but no compressive loading. Hence it may be feasible to use a lower strength alloy such as B722 and Ti-1828. The lower strength alloy is approach more ductile and should eliminate the chance of producing brittle blades in conditions.

► **Other Considerations**—I will think, with well defined, on such factors as load, compression differences in the bar, heating time and temperature, smoothness and cleanliness of lagging dies, amount of surface removal in die-casting, and grinding and finishing operations. No blade or other surface imperfections are tolerated and finishing marks must be removed as far as possible to eliminate sources of fatigue failure.

Cost of titanium blades is not so far out of line as that of lagging rings for wheels and casings. Material becomes a small fraction of the total cost because there are so many lagging operations on the blades. There is more for a large cut reduction. Fraser says by some process such as die casting.

The manufacturer the blade lagging



NEPTUNES MOVE ON TRACKS

Eleventh track on Lockheed Aircraft Corp.'s 72V Neptune lagging assembly line speeds sections around—one of the members can be moved from station to station in 1/2 in. as against a 3/4 in. period for the old method of rolling the heavy supporting dolly along the floor. The elevated rollers insure also persons accessibility to the entire assembly, cuts down on worker's walking because

tools and parts supplies are stored in bins adjacent to the line, makes more floor space available, simplifies utility services because quick disconnects for electricity and compressed air lines are adjacent to the tracks and allows more workers to be engaged in a single assembly at the same time. A similar system has been in use at Terren on Boeing B-47 rear fuselage assemblies.

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to produce. When good lagging are produced they will adequately replace steel blades as shown by many successful operational tests.

► **Part Aboard**—Every steel part is under consideration for titanium substitution where temperature permits, he says.

Alloys being developed are shift casings, compressor drive pins and bearing housings. These can be machined from titanium, but probably can benefit from lagging, he says—more efficient use of material can be reduced as well as mechanical property benefits from proper grain flow developed in the metal during lagging.

Titanium gains will require more successful load testing than has been produced to date, but there are several promising prospects. If a plant is used, there's no doubt it will have to be backed up by one of the better titanium alloys, Fraser says.

A bearing housing seemed to offer a simple, straightforward substitution of titanium for steel, but proved otherwise. The titanium couldn't be used in the die steel for steel lagging—concerns and was wouldn't fill and numerous jobs were produced, Fraser reports. And machining was difficult because of the high tool loads on this section.

But these problems are being solved and several more pounds will be due to from engines with titanium lagging used for these parts, he says.

Lines, titanium lagging upon the way to jet engines giving more thrust per pound of weight, but with constant strength effort for ductility, with strength, quality, practical fabrication and low cost.

McDonnell Tackles Engineer Shortage

McDonnell Aircraft Corp. and Wash. State University have joined in an effort to cure the critical shortage of engineers. A new agreement between the two groups that adequate courses in aeronautical sciences and related fields will be made available continuously at the university.

Another feature of the agreement provides for providing interest in providing students in graduate courses leading to master's and doctor's degrees. McDonnell engineers will be given the opportunity to make use of these courses to continue their professional training. Under the agreement, continue education of the university's ideas, and laboratories, as well as staff advisors, in engineering assistance, will be made available to McDonnell.

The new contract—running for three years—will be signed by McDonnell at a cost of \$27,000.



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Our latest press, measuring 40" x 120" x 66", with a capacity of over 1250 tons pressure, produces such parts as:

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PRODUCTION BRIEFING

►Mann Products Co., Inc., Los Angeles, has completed a \$1 million expansion at West Los Angeles to step up clamp, head and strap production.

►Marquardt Aircraft Co., Van Nuys, Calif., has built a large aircraft test facility at San Fernando Valley Airport under USAF contract.

►Midwest Tool & Engineering Co., Inc., Indianapolis aircraft products.com, has been acquired by E&B American Machine Co.

►Milford Buret & Machine Co., Milford, Conn., has purchased Pacific Buret & Machine Co., Alhambra, Calif., leaving to five the plants it owns.

► **Peterson Aviation Corp.**, Los Angeles, has acquired a factory building at 5140 W. Century Blvd., Los Angeles, to increase aircraft components output.

► Pratt & Whitney Aircraft division, United Aircraft Corp., E. Hartford, Conn., has started construction of a large addition to its service hangar at Rensselaer Airport, with completion scheduled for midsummer 1973.

► Precision Cast & Products, Inc., Paterson, N. J., has opened an 11,000 sq. ft. addition to its plant.

Willoughby Paint & Chemical Corp., Edgewater, L. I., N. Y., has acquired Titinate Corp., Union, N. J., longtime maker of aircraft finishes.

P. Kiser Aluminum & Chemical Corp., Newark, Ohio, has awarded a general contract for construction of its lagging facilities for the USAF heavy press program. Principal building and auxiliary structures will cover 380,000 sq ft. One lagging press, rated at 35,000 tons, will be nine stories tall. It and a 28,000-ton unit will be built by E. W. Bliss, Canton, Ohio.

► **Lincombe Airplane Corp.**, Garland, Tex., has started overhaul of Republic F-47 wings, landing gear, horizontal and vertical stabilizers, elevators and rudders under contract from Western Aircraft Corp.

• **Kohnsant Aviation, Inc.**, Teterboro, N. J., has established an engineering laboratory at its West Coast engineering office, Burbank, to test and evaluate vibration control equipment used in aerospace and electronics industries.



Thunderjet and lightning!

Ability of the USAF's new Republic F-84F Thunderjet to strike like lightning in support of ground troops—to carry an extra-heavy load of armament—and to fly exceptionally long distances as a super-fuel fighter—all reveal the importance of materials which provide maximum strength with minimum weight. And Republic Aviation Corporation, Farmingdale, Long Island, specifies Ducton Aircraft Tubing to meet these requirements.

Reasonable strength without weight character-

other plus specialized forming and machining capabilities make OTCU Tubing the choice of 26 leading U S plane manufacturers for landing gear, fuel lines, and many other applications.

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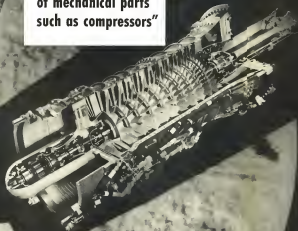
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SEAN O'NEILL, *Acquainted*, 223 p., \$24.95 • Chicago: Cross Currents Bk., 24 E. Wacker Dr., Chicago, 60601-1100. O'Neill's *Acquainted* is a memoir about his father, a man who was a part of the Chicago music scene. O'Neill's father was a part of the Chicago music scene. O'Neill's father was a part of the Chicago music scene.

TOMORROW'S AIRCRAFT:

What barriers remain?

**"Increased reliability
of mechanical parts
such as compressors"**



Westinghouse early recognized the need for extra reliability of compressor stationary diaphragms. Today, Westinghouse stands first for their contribution in stationary or static blade design and construction. How well they have developed a thoroughly dependable shrouded axial stationary blades on the compressor is dramatically demonstrated by Banchoff performance over Korea. For the fact is . . . there has not been a single operational loss of the Banshee (with Westinghouse J-34 turbojet engines) because of stator blade damage. Breakage at this point would be serious of course, ripping out all the other blades in the compressor. Westinghouse construction avoids a complete break; the blades may bend or hit, but won't tear out . . . both ends hold tight to the shroud.

Lighter, more durable jet engines, like the powerful, new J-40 which recently passed the Defense Department's grueling 150-hour qualification test, will aid our country's defense. Though other jet aircraft problems remain to be solved, Westinghouse axial-flow design, proved over Korea, points the way to the solution of future jet fighter and transport problems.

Westinghouse is investing millions of dollars and man-hours to help build American jet-propulsion leadership. Jet engines are produced at South Philadelphia and Kansas City plants by Westinghouse, America's Jet Engine Pioneer.

20002



Shroud shown is one half of the main rotor drive of a Westinghouse jet engine compressor. It carries 12 axial stationary diaphragms attached to main rotor drive by a pin and wedge bearing. The white lines above indicate one of the main rotor support struts.

THE SCOPE OF WESTINGHOUSE IN AVIATION

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Radio, Autopilot, Communication
Equipment and Electrical Systems

Ground equipment

Wind Tunnels, Airport Lighting,
Industrial Plant Apparatus

Aircraft systems components

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Gyroscopes, Temperature Control
Units, Servomotors, Equipment and
System Control, Circuit Breakers,
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 FARMER—C-119—DC-730
 GERSHBERG—F-8—DC-730
 JOCHIM—Convair—EC-802
 QUINN L. HART—Jaguar—DC-800
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AVIONICS

New Filter Slices Air Waves Finer

- Collins device makes more channels available.
- Mechanical unit solves this electronic problem.

By Halv Klav

A mechanical engineer began to work in electronics to help him solve his problems but it's obvious that the communications engineer calls on mechanical devices to solve his problems. However, a new mechanical filter developed by Collins Radio for ultra-narrow frequency (UF) radio circuits appears to reverse the picture.

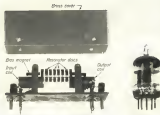
The purpose of a filter is to pass certain selected frequencies and reject all others. The new mechanical filter will allow the communications engineer to squeeze more channels into the already crowded radio spectrum by spacing them more closely in his transmitter. It will also allow him to design radio systems capable of rejecting unwanted adjacent channels.

Industry interest—There's lots of interest in the new Collins filter. Radio RCA, Los Angeles Laboratories, and Pan American Airways are reported to have purchased filters for experimentation. Collins has shown its confidence by using the mechanical filter in its new H-1400 Model 6188 high frequency airborne transmitter.

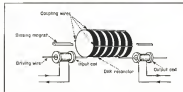
Collins thinks the mechanical filter will be extremely useful in wide-band (WB) communication equipment which since its birth are continuously operating in a possible solution to the problem of finding more radio channels in the crowded spectrum.

Here's what makes the mechanical filter an attractive:

- **Wide selection.** The mechanical filter has the nearly ideal rectangular shaped characteristics needed for interference when adjacent voice communication channels are very closely spaced.
- **Smaller.** In the 100 to 500 kc range, Collins uses its mechanical filter is smaller than any of the UF transducer networks used. And new mechanical filter produces fewer transducers.
- **Low loss.** Present production design units have losses below 28 db and compensated unit indicates that these can be cut to below 10 db in the future.
- **Fixed characteristics.** Once the me-



MECHANICAL FILTER compares in height with state-type tube (right).



BASIC OPERATING PRINCIPLES of new filter are illustrated by schematic.

chanical filter is constructed, its frequency characteristics are permanent and it needs no tuning. Collins says. The units are hermetically sealed so that coil aging and humidity, both of which affect UF transducers, are no problem for the mechanical filter, according to Collins.

How It Operates—The Collins filter converts its input signal into a mechanical vibration, then drops out unwanted vibration frequencies, and converts the passed vibration into an electrical output signal of the corresponding frequency.

The filter consists of eight nickel non alloy disks, two of which function only as end supports for the six center

resonators. All eight disks are connected to each other by means of three coupling wires welded to each disk. A nickel driving wire is attached to the second disk, and another to the seventh disk, first and last of the resonator disk. One driving wire "floats" inside the input coil, the other inside the output coil.

When a signal is applied to the input coil, its magnetic field causes the nickel driving wire to expand and contract (due to magnetostrictive effect) setting up longitudinal vibrations. These are transmitted to the input and resonator disk, and on to the other resonator disks through the coupling wires which act as springs. At the out-

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COLLINS mechanical filter has double bond gas characteristics

put end, vibration of the during use induces an electrical current in the output coil by the reverse process. Small permanent magnets near the input and output coils serve to establish a housing magnetic field.

Small micro transducers are attached across the input and output coils to provide a low Q resonant circuit. In put and output coil impedance is 5,000 ohms, which allows the filter to be directly connected to vacuum tube plate and grid circuits. The complete filter is housed in a hermetically sealed brass case, which measures about 1 1/2 x 1 1/2 x 1 1/2 in.

Filter Characteristics—The filter, which Collins has recently producing, has a center frequency of 455 Kc. However, Collins says that unmodulated filters can be built and should show as favorable. An air frequency in the 100 to 100 Kc range. Below 100 Kc, the filter use gets objectionably big; higher frequencies the fabrication of two elements becomes a problem.

Collins has built an 455 Kc filter with a 1/2 in. diameter of 1 to 5 Kc in a 1/2 in. diameter brass. A 1/2 in. diameter filter is now in pilot production and Collins expects to be producing the 1 and 5 Kc units soon. Other characteristics of the pilot production 455 Kc filter are:

- Peak to valley ratio: 1 db in gain has 1 experimental run have been built with 1/2 db ratio.
- Frequency loss: 25 db (which can eventually be improved by better one plane between the coils and driving signal).
- Overload input power: 0.015 watt.
- Time delay: 1 to 1 millisecond in type, best.
- Operating temperature range: 10C to 200C (outlet temperature compensates from).
- Meets ANE-19 vibration repair needs.

Collins selected the filter to filter

Vibration Engineering that solves your problems

PROBLEM: To prevent transportation damage to packaged airplane engines.

SOLUTION: Inmate's Shock Mounts.



So shock mounts are low pressure plates are now patented in many and one all. They're used as an excellent example of good product engineering that makes certain measures of engine to guard them against shock and vibration in shipment.

The shock mount Inmate's Shock Mount was developed for this job. It was engineered to combine low cost design with high deflection capacity in order to provide both good absorption of shock and dependable support. With these shock mounts are protected from damage while in transit. In service, too.

While this patent, in a special case of vibration engineering, it shows what it takes to deal effectively with vibration problems—namely, a company that has a good record for solving problems in isolation, restraint, suspension, vibration and measurement of vibration. Write us. Bulletin available.

Circle 4 on Card 10

The entire 1945 Patent No. 2,416,111 is available for

**THE MB
MANUFACTURING COMPANY, Inc.**
1067 State Street, New Haven 11, Conn.

tion in the 10 to 15 cps range while passing a 455 Kc carrier through the filter to a low frequency oscillator. An oscillator-induced modulation would then show up in the receiver, but would not. Collins reports. Tests of filter characteristics before and after the vibration tests showed no change due to vibration.

Easy to Produce, Easy to Service

First Instrument Co. (division of Sperry Corp.) uses its pioneering techniques in packaging the atomic components for an aircraft computer which it builds for the Navy. The Ford procedure simplifies its own manufacturing and assembly and the Navy's maintenance and replacement problems.

Subassemblies are constructed on pre-punched standard laminated boards. The simplified wiring and soldering operations and punch card layout is in completely word before installation. Once the laminated board is used, it is served to an L-shaped chassis.

Tube sockets and a quick-disconnect



PRE-WIRED assembly components are



MOUNTED on chassis, which is then



INSTALLED with other subassemblies

Carter pumps give performance plus



North America's P-51-D

Sabrejet is a most formidable atmosphere in the hands of Air Force pilots noted for spring steel grade, infinite skill, and ingenuity. The General Electric jet engine gets a big power boost through the use of an afterburner. The performance and agility of this subsonic fighter hinges on the ability of a tiny 4-point pump to unfurling deliver a large volume of fuel under extremely high pressure. A Carter designed pump does this job.

OVERHEAD ENGINE: Once this plane drops its nose the belly of its undercarriage, the horizontal tank of supplying fuel and liquid mixture to the four engines (not shown) is exposed to the air. The Carter engine controls pump.



REAR GAS EXHAUST: One of the problems of inverted flight was solved when in-flight refueling was proved practical and safe. Carter designed and manufactured pumps for the successful transfer of fuel from the aerial tankers to wing of today's aircraft.

OTHER CARTER ACHIEVEMENTS

The Carter trademark is increasingly apparent where special-purpose fuel valves are demanded. Typical developments are: Fuel pressure limiter, fuel tank inlet valves, pressure limiting devices, fuel flow limiters.

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type connector for slip underneath and are quickly secured to the chassis shell. Tubes and transducers are added and the transducer tape soldered. The chassis is then ready to join other sub-assemblies to the main assembly.

Individual chassis are mounted on the base of a T-frame frame, by sliding them between the rails and tightening a jack across which support the chassis. The frame connector on the chassis marks with a scale connector on the T-frame. The T-frame shell fits inside of a case which can be filled with parts and can be pressurized up to 20 psi.

30000 FILTER CENTER 30000

• **RCA To Build Hughes Fire Control.** The USAF is setting up the Radio Corp. of America in a second source to build the Hughes Aircraft fire control system and on current crop of interceptors (F-94C, F-94D, F-94E). Navy also is reported to be using RCA as a second source for its Westinghouse interceptors fire control system.

• **Ashtek For Radar Designers—General Electric** has developed a circular slide rule to enable it also to determine the maximum range of pulse-type radar when radar system design parameters are known. The GE device serves the radars calculation involving operational power, of seven variables (pulse duration, repetition rate, peak power, antenna gain, wavelength, receiver sensitivity and target size). The circular slide rule should be useful in determining the effect of a change in one or more system variables on radar performance. Reverse side of the slide rule contains other scales, including one for

calculating antenna gain. The new rule may be purchased for \$7.50 from General Electric, Connecticut and General Electric, Electronics Park, Syracuse, N. Y.

• **New Collins Course Indicator—Collins Radio** is fighting testing a new 7-in. diameter course indicator as a possible replacement for the larger (7-in.) indicator now used in its Integrated Flight System. Course indicator gives a graphic presentation of plane's position relative to surroundings or its location below.

• **New Technical Bulletin.** Specialty transformers for avionics equipment are described in recent brochure published by Collins Electric & Mfg. Co. The publication lists various engineered transformers for radar, gyro, or communications equipment in either open-core or hermetically sealed types which meet military specifications (Dept. AT, 2121 West Ohio St., Burbank, Calif.).

• **Stallboard sports crystals** for communications equipment and other applications are described in a new catalog issued by James Knight Co., Sandusky, N. Y. More than 45 different crystals are listed.

• **New electronic capacities** of the transistor type which meet JAN C-5 are described in catalog No. 17 now available from Spangco Electric Co. (127 Main St., North Adams, Mass.).

• **Antenna and ground interconnecting component**, including plotting and table-reading devices, are described in a four-page brochure by the Applied Science Corp. of Princeton. Looking block at antennas which can be assembled into a system to meet individual requirements see listed (P. O. Box 44, Princeton, N. J.).



KEDIPON BUILDS CORNET SIMULATOR

Remote activities in the flight simulator field, as evidenced by the Cornet jet base unit, have gotten a big boost from a \$1-million order by the Canadian Government

to build F-86 fighter trainers. The order goes to Rockwell, Ltd. which previously built a Boeing Stearman simulator, under license to Chance Vought, for RCNAC.



IMPORTANT
ANNOUNCEMENT

...to Engineers
and Scientists

You can now fill vital positions
in our guided missile projects

Chance Vought Aircraft, a supplier of high performance Navy aircraft for 35 years, is presently engaged in highly classified work on guided missiles under Navy contract. These missiles are in restricted production for intensive experimental use. They are flying and their performance has been excellent.

Engineering and scientific personnel with backgrounds in Aerodynamics or Electronics will find exceptional opportunities for employment on these interesting projects. Openings are available to personnel with Ph.D. and M.S. degrees, or B.S. degrees with related missile experience.

For further information write Engineering Personnel Section, Chance Vought Aircraft, P. O. Box 9927, Dallas, Texas.



CHANCE VOUCHT AIRCRAFT

Divisions of United Aircraft Corporation
DALLAS, TEXAS

EQUIPMENT



EXPERIMENTAL fuel burner pump equipment at Hydro-Aire is part of...

New Setup to Test Air Accessories

Hydro-Aire's new facility is designed to "fabricate, evaluate, qualify" equipment for aviation firms.

William J. Conklin

Burbank—New test and research facilities for fabrication, development and qualification of aircraft accessories have become available to the aviation industry with the opening of Hydro-Aire, Inc.'s new main plant here.

This new plant covers 78,800 sq. ft. and will up production capacity 400%. The expanding research laboratory, expanded to 7,200 sq. ft., will test fuel, hydraulic, pneumatic and electrical aircraft accessory equipment. The lab is nearing completion when reported recently by Avionics Week. It houses \$500,000 worth of test equipment, is staffed by 18 engineers, technicians and mechanics.

► **Longshore Extension**—"Given a preliminary design, the laboratory is equipped to fabricate a prototype part, evaluate the performance under conditions simulating actual operation, and to qualify the test standing in customer and military specifications," says O. A. Wright, chief design and aircraft engineer.

A liquid carbon-dioxide tank feeds a refrigeration circuit at all sections of the laboratory. This "dry ice" can be hauled from the coast warehouse in liquid form and used through spray nozzles for cooling such items as temperature-sensitive components. A walk-in cold chamber with a 100 sq. ft. working area will take test temperatures down as low as -104°F. Sufficient space available

here will simulate altitudes up to 65,000 ft. One large chamber capable of maintaining regulated temperatures up to 600°F is used for high heat testing.

► **Fuel Section Facilities**—New equipment now being installed will permit simulation of actual aircraft fueling conditions.

Three test flow circuits are used to test accessories in the fuel laboratory. Components such as large gate valves or pumps flow values are tested on a large circuit with a variable flow up to 660 gpm and shut-off pressure of 450 psi. This circuit consists of a single-stage centrifugal pump with a variable speed drive, a 1,400 gal. reservoir, and a hot and cold heat exchanger.

The second circuit, which includes a two-stage centrifugal pump and a 200 gal. reservoir, is used for testing of accessories as part of a complete engine. It provides a variable flow up to 200 gpm with a shut-off pressure of 250 psi. It also is connected to a separate fuel tank.

The third flow circuit is part of a second test bench, which has a hand-operated pump for high and low flow tests, and accommodates a flow of 70 gpm and a shut-off pressure of 70 psi.

The test stands on the fuel area are built to handle both submerged fuel pumps and engine driven pumps. The test stands are connected to an altitude system capable of a climb to a simulated altitude of 16,000 ft. in one minute.

► **Phased Area—One third** of the laboratory is set aside for the testing of pneumatically operated accessories. It consists of a test room and an adjacent compressor room. Heavy machinery in the compressor room is capable of supplying air at 1.5 lb. per sq. in. at 135 psi. Other machinery will provide air at pressures and flows up to 3,000 psi and 45 cfm free air. A gas-fired heater can heat the air to 800°F and the air also can be continuously and electrically dried. The supply of hot and cold air is piped into the test room to a system with several outlets to allow for more than one test setup at a time.

► **Other Check Work—Hydraulic** testing is done on a large test bench, which includes a piston pump capable of flows up to 25 gpm and pressures up to 5,000 psi.

Hydraulic test setup includes a power supply with several 4-cu. gallon oil tanks, adjustable from 10 to 12, and capable of supplying 300 gpm. A three-phase 10-hp alternator can produce e.c. from 187 to 590 cps at voltages from 100 to 250.

Factor on the vibration mechanism has a capacity of 25 lb. through a frequency range from 4 to 500 cps, 18 lb. up to 1,000 cps and a constant frequency of 75,000 cps. It thus will test accessories in accordance with requirements of AF Spec. 41065B and MIL-E-5272.

The research lab has its own machine shop to reduce maintenance with production facilities.

The laboratory has recently acquired a new trace recording oscillograph with a carrier amplifier for recording temperature, pressure and electrical transients.

Ted Scott, head of the testing department, hopes to add by the end of the year fuel separator test equipment and a 100-hp, 10,000-cps dynamometer as well as salt spray and humidity test equipment.

Unit Tests Plane Circuit Breakers

A portable tester for checking out circuit breakers in aircraft is one of several new pieces of equipment recently announced by Greer Hydraulics.

This new model, CCB-13, has been ordered by A. V. Roe, Canada, Ltd., the firm reports. It indicates how long a circuit breaker takes to trip from an overload to protect the aircraft. It will test breakers rated up to 150 amp at 28 v. d.c.

The equipment includes a voltmeter, a dual-scale ammeter (0-80 amp and 0-150 amp), negative (pilot) manual ports, 13 locking switches to tie the breaker, and a 1,000-sec. manual reset timer.

► **Accessories K20—Greer** also is introduced

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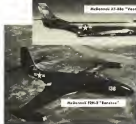
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...in many of today's leading aircraft. For example, Sier-Bath supplies gear components for the famous Waddinghouse J34 Jet Engines... which power the two modern McDonnell Aircraft Corporation planes shown at left.

McDonnell RF-44 "Phantom"



McDonnell F2H-2 "Banshee"

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Remotely Operated Switches for Airframe and
and 100 lb. Breakdown Switches (MELETRON)
for Aircraft Repair Service, Ground, in use since
1947 Boeing Aircraft Corp., Long Beach 25, Calif.

reducing aligned with the damping
coils line and is quickly attached with
another nut. Maximum visual observa-
tion (viewed by looking over, wheel
door, etc.) is a 4-deg. angle. Height of
the attachment bracket is adjust-
able, making the instrument readable
whether directed to the low belly of a
4-4 or high underside of a Comet.

Use of the potentiometer compen-
sating is not new, but previous in-
stallations were mounted on top of the
aircraft. EAL found many drawbacks
to this method, including: inaccuracy
due to the operator looking upwards in
his previous pitch stop the plane.

• **What it is:** The potentiometer has the
quality of being straightforward, use-
ful and simple. The potentiometer
is a 10 in. drafting potentiometer calibrated
in 1 in. increments through 360 deg.
Readings of 1 deg. may be interpreted.
The sight is a Maytag 2M4-D, 4K
sight scope. A battery powered "beam
gun" built as a penlight size illuminates
telescope reticle for night work.

• **Putting it to use:** Installation of the
potentiometer takes less than five minutes,
says EAL, and first proof alignment
eliminates possibility of personnel error.

After adjusting potentiometer to his height,
the operator sights the instrument on
groundmark, easily recognizable ob-
jects, all over one aerial tank distant.
The bearing of each object has been
determined by celestial assistants and
calculated to minutes of an arc.

The operator enters the code of the
aircraft's heading through various
connections of relay and other systems.

Enders says, "The accuracy of head-
ing of 1 deg. far exceeds the calibra-
tion limitation of the line gate system."

Hot Air for Testing

A tag that heats high-pressure air in
other gases to be used in testing com-
ponents, has attracted the interest of
jet engine manufacturers, according to
the maker of the device, Thermal Re-
search & Engineering Corp.

The unit, gas or oil-fired, delivers air
at a test pressure of about 1,200 psi with
operating temperatures up to 1,600 deg.
F at 15 gpm, and the company
claims it can be made to go higher.
The heat exchanger is sold as a com-
plete package, including burner (with
or without controls).

The equipment comes in two ver-
sions—Model 1019 weighs 850 lb.,
provides 775,000 Btu/hr., Model 1019
weighs 2,500 lb., puts out 2,500,000
Btu/hr. The units are new high-
pressure burners, and to give extremely
high heat release in maximum combus-
tion space and provide products of com-
bustion at maximum velocities and
temperatures.

Thermal Research & Engineering
Corp., Coocheshaw, Pa.



Vickers Model A-13200-B
AN-8279-ACD



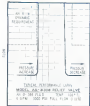
Vickers Model A-13200-B
AN-8279-ACD



Vickers Model A-13100-B
AN-8279-ACD



Pressure variation from cranking point to
maximum rated capacity of Vickers Two-Port
Balanced Piston Relief Valve is considerably
less than permitted under Specification
MIL-V-5523. Consequently less pressure dif-
ference is required between rated valve setting
and unloading valve pressure.



Graph showing pressure less internal leakage
of Vickers Two-Port Balanced Piston Relief
Valve

These **VICKERS** RELIEF VALVES TWO PORT • BALANCED PISTON Conform to Specification MIL-V-5523

The Vickers Two-Port Balanced Piston Relief Valves illustrated here con-
form to Specification MIL-V-5523. Their rated capacities (2, 5 and 9 gpm)
are greater than required by this Specification (1.5, 4.5 and 6 gpm re-
spectively).

The curves at the left illustrate two important characteristics of these
valves: (1) very low pressure variation from cranking point to maximum
rated capacity, and (2) unusually low internal leakage (less than required
by Specification MIL-V-5523). Swasther operation and greater accuracy
throughout a wide range of pressure adjustment are other significant
advantages. Operating pressure ranges adjustable from 500 to 4500 psi
without parts change.

These valves are also available in four-port models and can be provided
with a vent control for unloading the system pressure. For further in-
formation about the complete line of Vickers Balanced Piston Relief
Valves write for New Bulletin A-3209.

VICKERS Incorporated

DIVISION OF THE SPERRY CORPORATION
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ENGINEERS AND SUPPLIERS OF THE MARINE EQUIPMENT SINCE 1921

OFF THE LINE

These World Airlines has seen into a starter at its engine, test stands at Keesler City. Believed good from newly tested jet engine test cells get sorted into the categories as made of T.O.A.'s engine being run in and tested. Circulator air temperatures as high as 115F are what have given the airline trouble.

Pacific Aerospace Corp. will overhaul and convert all of the American World Airways (AWA) R4160 T581G engines under a contract that runs to November 1975, the engine agency announced. The overhaul firm will modernize the engines from the T581G to the B6 configuration and do any work required by the AWA's detailed engine overhaul specifications. PAC says it is the only overhaul engine maintenance facility equipped to overhaul the Suburban engine.

Aircraft engine testing is a major test and source of maintenance to many airlines. Newbedco is experimenting with a new product that reportedly reduces engine test heat by absorption, prevents icing, frequently the material and allows certain to absorb their "drips" during reclamation. The product, called "Reflux," is said to be aluminum in composition from which condensation on the engine fabric to form a protective coating. Producer is Macdonald Engineering Co.

Environmental test chambers for transportation hardware, tests on aircraft and other components have been standardized into five basic models by Tenney Engineering, Inc., 26 Ave. B, Newark, N. J.



HEATER PUMP-MOTOR

Jetcraft combustion heaters are supplied with the pump-motor combination, known as RD-9546, rated to deliver high pressure aviation gas at 55 gph. Discharge pressure is variable from sea level to 40,000 ft. altitude. Pump weighs 3.4 lb. and measures 7 1/2 in. long. Research Engineers, Inc., Inc., Elgin, Ill.

AVIATION WEEK, November 3, 1972

NEW Pressure Switches

FOR AIRCRAFT APPLICATIONS OF EVERY KIND

New Manning, Maxwell & Moore makes available to you new pressure switches in three basic designs. Regular production units of these precision-built pressure switches conform strictly to recognized engineering performance standards and pass the toughest specifications of the U.S.A.F. They are adaptable to any aircraft application, and include special types for individual needs.



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High static pressure gauge or differential pressure switch . . . single pole, double throw.



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Low static pressure gauge or differential pressure switch . . . single pole, double throw.



FOR ROCKET INSTALLATION

Reversibly acted high static pressure gauge pressure switch . . . single pole, double throw.

We welcome the opportunity to study your specific needs within the field of aircraft pressure switch applications. We are fully equipped to run exhaustive environmental seal vibration tests in complete accordance with the requirements for medium high-speed aircraft at every type. For prompt attention to your inquiry, please write our Aircraft Engineering Department at the address below.

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MODEL 10401



MODEL 10402
10410



MODEL 10409



MODEL 10408



MODEL 10412



MODEL 10411



MODEL 10414



MODEL 10415

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ARO Two-Stage Automatic Condensor-Flow Oxygen Regulators are precision-made to provide better performance . . . simplified servicing.

All models are variants of a basic regulator, Model 10409, and will give specified performance on inlet pressures of 50-2000 p.s.i. These models cover all currently known installation requirements. Models can be furnished with output performance according to Civil Aeronautics or Type A-11 specifications.

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MODEL 10407



MODEL 10411



MODEL 10408



MODEL 10414



MODEL 10409

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with aluminum elbow fittings



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Machined from solid forgings of higher strength aluminum, Resistoflex hose fittings resist vibration and fatigue. Their true bore and smooth interior finish promote full flow with less turbulence or pressure drop. Elbows are compact—fit into tight spots with ample wrench clearance.

See dimensional and test data in the Resistoflex Aircraft Catalog—write and tell us how many copies you want.

Engine Designers Plan to use Resistoflex fittings that now carry SAE and "BuMIL" approval

RESISTOFLEX

CORPORATION
Bulldville 9, New Jersey

NEW AVIATION PRODUCTS



Koldweld hand tool developed by Utica.



Mechanism of Koldweld arc action.

Firms to Boost Koldweld Process

The Koldweld process, a low-cost method of welding nonferrous metals with pressure alone, has secured support from Utica Dens Forge & Tool Corp.

The firm has made arrangements with Koldweld Corp. to develop and supply tools for the process. The plan covers development and sale of large production machines, as well as hand tools of study developed by Utica.

Koldweld Corp. is the only licensee in the U. S. The process is owned by General Electric Co., Ltd., England (no relation to the American firm). It was brought here two years ago by Koldweld's president, William Doherty, a founder of Corrad-Doherty Corp.

At present, parts are cold-pressure-welded by special hand press. Utica expects to lighten and streamline the now cumbersome tool for everyday use by workers and homeowners. Koldwelding already has been used in the aircraft industry by Ryan Aircraft Inc. in joining two sections of a metal fuel tank. Koldweld and Utica are now negotiating



INFORMATION ON POSITIONS AT NORTHROP

Northrop Aircraft, Inc. is engaged in vital important projects in scientific and engineering development, in addition to aircraft production. The program is diversified, interesting and long-range. Exceptional opportunities exist for qualified individuals.

The most responsible positions will go to top-flight engineers and scientists. However, a number of excellent positions exist for capable, but less experienced, engineers. Some examples of the types of positions now open are:

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ELECTRONIC PROJECT ENGINEER...

ELECTRONIC INSTRUMENTATION

ENGINEER...

FLIGHT-TEST ENGINEER...

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ELECTRO-MECHANICAL DESIGNER...

ELECTRONIC INSTALLATION

DESIGNER...

Qualified engineers and scientists who wish to locate permanently in Southern California are invited to write for further information regarding these interesting, long-range positions. Please include an outline of your experience and training.

Allowance for travel expenses.

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ON GUARD!

Guardians of the upper reaches of the Western Air Defense Command are the men of the 84th Squadron, at Hamilton Air Force Base. The 84th flies the U. S. Air Force's new all-weather interceptors—fast, deadly Northrop F-89 Scorpions.



Northrop Aircraft, Inc.
Hawthorne, California

Pioneer Builders of Night and All-Weather Fighters



Business is going up!

Busy executives are taking to the air.

More and more of them are flying their own planes or are using company planes to go places faster, more comfortably, entirely free of schedules.

And in more than 800 airports they look to the Esso Wings for quality and dependability in aviation products and service.

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other applications of the method and see many possibilities in the aviation field.

Only hand pressure is used, no flame, flux, or other chemical is applied. Yet the weld is said to have greater tensile strength in some applications than the metal itself. Pressure and the manner in which it is exerted by the plier heads causes an intermolecular flow between parts being joined, microphotographs show. The pliers can be used after short annealing, it is claimed.

Removal of the oxide film on the metal surfaces is required before the welding operation. This can be accomplished with a wire brush.

Ultra-sonic hand tools for welding wire and for metal sheets seem well be available. These and later tool developments will permit "welds" of uniform confidence at unprecedented low cost.

Kobalicki states:

This method has been used to weld terminal lugs to bonding cables of the type used in aircraft. In both metal sheet and wire and electrical conductors, by joining the sheets and for assembling of tubes. With tubes, the joint can be made virtually flush with one the connection in the wire gap in the wire itself after a small flake of metal at the bond has been removed. Ultra-sonic a number of companies in England are adopting this method of connecting cables for its speed, reduced radio interference through improved connections, and elimination of discharges caused by foreign matter. The butt weld is made in seconds.

Ultra-sonic and Kobalicki believe the process will prove a boon in welding copper leads to aluminum chassis. Small copper leads can be butt-welded to the aluminum chassis, then the copper wire welded to the disk.



SOLENOID VALVE

Re-designed F-84 fuel system incorporates the solenoid valve made by Valco Engineering Corp., Newark, N. J. This serves as pilot valve and has exclusive floating shut-off, providing reliable operation under extreme back pressure and minimum pressure drop, according to firm.

SPECIAL ATTENTION!

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If the harmful effects of oil, grease and paint, which cling and with in insulating materials, be sure to insulate these materials. They are damaged and subjected to an insulating defect, pointing out their advantages and how to apply them of wire and cable insulation materials.

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STANDARD CLAMPS FOR SPECIAL APPLICATIONS

Fast Camera Speeds Engine-Knock Study

A research camera so fast its mirror clears it would go through a million frames if required for 10 sec. is being used to study of the causes for knock in spark-ignited piston engines.

The study is sponsored jointly by the Navy Bureau of Aeronautics, the aircraft, automotive and oil industries. It was disclosed at the recent International Symposium on High-Speed Photography. Details of the unit, known as an "ultrafast" camera, were given in a talk prepared by C. D. Miller and Arthur Schatz of Battelle Institute, independent applied research organization.

The camera also can be used to record high-speed phenomena associated with helicopter and jet engine operation and for a variety of other tests. It is said to be about ten times faster than other high-speed cameras commercially available and was developed to fill the need for equipment in the speed range of 20,000 to 100,000 frames/sec.

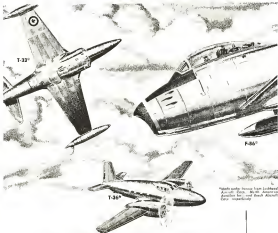
The ultrafast is a faster model of a design developed by Miller when he was associated with the National Advisory Committee for Aeronautics. Photographs are said to be comparable in clarity and detail with those obtained in cameras operating at much lower speeds. Distortion of objects from frame to frame is avoided because pictures can be taken through the same lens from the same viewpoint, according to Miller. A single area of 500 frames can be projected as a motion picture immediately after development of the film without reprinting and re-exposing the frames.

Canada has only one moving part.

ALSO ON THE MARKET

Flexible X-ray of 1 million volt capacity and weighing 150 lb can report steel 71 in. thick, too great to poke inside of casting for complete going over. Bendless window of X-ray tube permits inspection of lighter metals by allowing escape of softer, less-penetrating X-rays from tube. General Electric Co., 4855 Electric Ave., Milwaukee 14.

Minibest test point jack has new protective collar around conductor to permit safe circuit voltage readings up to 1000 v. ac. from the front of equipment panels. Collar protects personnel from flashover during high voltage checks. Made to J4N-F14, MTS-23 and MIL-P-144-CFG specifications by Allen Products Co., 117 N. Main St., Rockton, Mass.



Single engine trainer from Lockheed Aircraft Corp. With design as flexible as... and fresh aircraft... technology.

What are we building at Canadair?

Canadair is building high speed "T-33" jet trainers and "F-86" fighter interceptors... while accelerated production of F-86 Sabre jet fighters continues unabated.

In the production of three types of aircraft, Canadair exhibits the high flexibility of this engineering plant. It is this ability to handle varied assignments simultaneously that has assured such interest in military and civil aviation circles.

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Circle 100

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Setting of 22-1/2" High Temp. Dry Flood for Aluminum Heat Treating, 20-2500 lbs. capacity



Setting of 22-1/2" High Temp. Dry Flood for Aluminum Heat Treating, 20-2500 lbs. capacity

Wherever the solution heat treating of aluminum plays a prominent part in production operations, more and more plants are turning to DESPATCH heat treatment for better uniformity, speed-of-quench and economy.

The DESPATCH bottom entry, quench-quench furnace shown, is being used in the aircraft division of a large West Coast manufacturing firm for the solution heat treating of aluminum aircraft parts. Developed by DESPATCH engineers especially to meet rigid Government standards and airframe specifications, this furnace is one of several that have been designed, built and installed by DESPATCH for major plants throughout the country, now engaged in Defense Production.

RAPID QUENCH—less than 10 seconds. Electrically heated, the furnace has a temperature uniformity within $\pm 5^\circ\text{F}$, and a temperature range up to 1240°F . The time consumed from work chamber to quench jet is less than 10 seconds. Doors and elevators are remote-controlled and air-operated. Capacity of work chamber is 600 lbs. of aluminum plus supporting steel.

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DESPATCH engineers are ready to talk things over with you, offer advice, or design, build and install the proper equipment for your plant. There is a resident engineer near you, who will assist with your heat treating requirements.

FIFTIETH ANNIVERSARY

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AIR TRANSPORT

BOAC Says Comets Now Pay Own Way

- First four months of new service net profit.
- Problems: speedup of ground, operating pace.

By Robert Elms

London—Courtesy to some Airwayman opinion, British Overseas Airways Corp. is spending its first Comet jet transport service at a profit.

BOAC treasurer Basil Smalgrave told AVIATION WEEK that the first four months of Comet operations—carrying 4,576 ton-paying passengers on the London-Edinburgh-Springbok route—resulted in a net profit of \$36,500.

"This is not simply an operating profit," Smalgrave emphasized. "It was earned after the Comet had had time to pay off the cost of the corporation's ground and commercial overhead and after taking its share of the whole corporation's cost of Comet development. Comet development cost, including route development, are borne by the Comet and alone and are not spread over the whole corporation."

BOAC's Comet development costs are being amortized over the life of the aircraft—estimated at about eight years. BOAC also expects that the Comet's cost per capacity ton-mile is running higher than similarly because extra fuel is being burned as an additional safety provision with a new type aircraft.

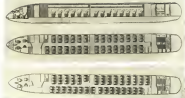
"We are flying with better jet, expanded and better crew," explains BOAC's chairman Sir Miles Thomas. "Soon, we will be flying with just better."

Land Factor.—For the Springbok route, BOAC calculates the break-even load factor for the 36-passenger Comet 1 at 72.5%. During the first four months of operation (May-August) the actual load factor was 79.1% with a rate to 83.5% in the early fall. Although bare, advance reservations indicate that high load factors will be maintained at least through next spring.

The BOAC Comet was handled by Capt. M. J. W. Alderson, but acknowledged several thousand hours of jet transport time and is now flying its new Comet 1 at a rate of \$50 for a month. Capt. Alderson told AVIATION WEEK the timetable for the Comet was a lot less than that required in Statesman and Constellation but



BOAC COMET (Series 1 model shown). Crews learn quicker jet flying rules now.



SERIES 1 COMET (top) will seat 36 passengers equipped with 15 seats, high freight version (below) will have extra layout for 75.

note that the Comet took a good deal longer. BOAC now has some 18 jets scheduled out on Comets and a training flight program for Pan American Airways, Canadian Pacific, RCAF, British Commonwealth Pacific Airlines and the French UAT. BOAC pilots report a marked preference for the Comet, claiming it is simpler to operate than a piston-powered transport and much less fatiguing.

BOAC has found the Comet crew training and route familiarization more expensive than had been anticipated. In addition to the regular Comet service, BOAC is using one Comet for route familiarization and another for transition training.

Some commercial airlines flying are made with Comets, including flying the Delta of Edinburgh to and from

the Olympic arena held at Helsinki. The BOAC Comet seat put in more than 1,000 hr. of route flying before exchanging passenger service for the Springbok route.

Budget defense in Comet flying and conventional transport operations is the increased loss of both light and ground operations and the rapid pre-planned light pattern required for more conventional operations.

Here, too, some of the things the BOAC Comet crew has learned about jet transport operations.

Engine.—Civil version of the Comet 1A01 used in the Comet is now operating at a 210 hr. interval between overhauls. This is expected to increase shortly to 375 hr. and eventually to 450 hr. De Havilland is now doing all Comet overhauls but BOAC overhauls

plans to do its own jet overhaul work at its maintenance base in Wales. BOAC now makes its own modifications, then has airplanes built 175 by. An example of two recent engine replacement on each respective.

Most of the upshot lines are repairable for further use. A spare engine is stored at each station along the Concorde routes.

► **Flight Operations**—Takeoff is made after an engine warm-up to full rpm, holding the aircraft with the brakes. The brakes are released and the Concorde allowed to build up speed to a range of 15% above stalling speed. This is con-

sidered a satisfactory initial safety speed by the BOAC Concorde unit.

Experience with three-engine turbofans indicates no winging tendency of the aircraft even when an outboard engine is cut off. The disc governing of the engine maintains asymmetrical power loads. Lack of a prop at the job eliminates drag generated from windmilling or tailwind prop on taking piston engines.

The British has designed a duct that catches air on both sides leading gas wheel wells when the gas is down, thus eliminating drag from an extended door and also increasing the efficiency



BOAC'S CAPT. ANDERSON. Photo given by the Concorde because of its simplicity.

of the wing during the critical periods of landing and takeoff. At landing weight the Concorde can maintain a 1,100 ft per min rate of climb on three engines with which and flaps up.

The large amount of fuel burned off en route brings the wing loading of the Concorde down to about 18 lb. per sq. ft. during approach and touchdown. This low wing loading, at landing weight, makes it possible to use approach and touchdown speeds slower than those normally employed by a Stratocruiser or Constellation.

Loss of an engine on route seems to be a "built down" procedure to the altitude where three-engine cruise can be maintained. Operational calculations on a 1,500 nm stage length with a "drift-down" to 37,000 ft from a normal landing engine failure at the half-way point show a fuel penalty equal to 12% of full payload can be expected. A similar calculation with the three-engine cruise executed at 15,000 ft would result in a fuel penalty equal to 64% of full payload.

Although the Concorde's climb and descent through cruise conditions is so rapid that its thermal deicing system requires little, which, says, flight experts have shown that it would be an absolute necessity in one of three-engine cruise in tropical conditions. Being a "drift down" to altitudes where heavy icing is a problem.

Normal cruising at about between 36,000 and 40,000 ft. Concorde pilots have been on instruments at 40,000 ft and an altitude have gone to 45,000 ft to avoid weather. Concorde pilots report cloud tops are generally much higher than forecast and they therefore consider safety clearances at 40,000 ft.

The effect of higher temperatures on en route cruising operations has been such less than was anticipated but the temperature at takeoff points has an

important effect on maximum gross loads. Above the temperature limitations recently calculated for Concorde operations, each additional degree centigrade at takeoff point reduces the Concorde gross by the weight of a passenger and his baggage. During cruising conditions an increase of 7 deg. C is expected to cost a similar weight penalty.

► **Navigation**—Basic navigation is done by radio with celestial procedures used as a backup in case of radio failure. Concorde pilots feel that a universal time-guaranteed aid system is needed for global jet operations both for en route and terminal procedures. More VHF homing beacons, distance measuring equipment with range up to 100 nm, and more automatic landing systems are needed along present and future jet transport routes.

► **Weather**—Forecast forecasts on surface winds and temperatures at several points and good high-level wind forecasts are most necessary for jet operations. Prompter information on deteriorating weather conditions at the approach is a must since the further out a Concorde can be diverted to an alternate, the more economically a flight path can be planned to the alternate.

Little else as turbulence has been encountered in Concorde operations to date. BOAC meteorologists survey meteorologists include 24-hr forecasts on the location of high-altitude jet streams as these can be an important factor in planning flight operations. The jet streams vary considerably in width but they are seldom more than 5,000 feet deep.

► **Traffic Control**—Actual flight operations have proved high-altitude stacking often as practical advantages in fuel consumption over a normal traffic pattern descent. Since the Concorde usually begins its takeoff about 200 nm from destination it should be directed to an alternate from this point if terminal weather has gone sour.

Concorde air for an aerial landing priority in the Heathrow Airport traffic pattern over London. A Concorde has stacked as long as an hour over London and has diverted to its first way at Heathrow (300 nm) where all of its southern England alternates were fogged-in and

Stacking technique calls for two engines throttled back almost to idling and the other two maintained at high rpm. Use of air brakes and the flexible shock absorbers system rule, at speeds of 3,000 ft per minute operation-

sily standard under instrument conditions.

► **Ground Handling Problems**—The Concorde can be loaded actively but to avoid excessive fuel consumption and avoid being quickly into its ramp station. The jet strikes and bounces are 7 ft off the ground and do not present a hazard to people on the ground even at full engine output. It now takes about 40 min from door-open to door-closed to complete terminal operations at several refueling stops but customs, immigration and police hardly affect overhead to lighten actual ground time well beyond the technical requirements.

From a technical viewpoint the BOAC Concorde pilots feel that they are now operating over too short stage lengths to get the most out of a jet transport. The shortest leg is the 69 nm between Livingston and Johannesburg and the longest 1,560 nm from Beirut to Khartoum.

The Concorde pilots feel that as more experience is gained a good deal more operational flexibility will appear in the new-age pattern of flight operations. None of the pilots should not on the Concorde have expanded any desire to return to piston-powered transports.



LOCKHEED RTV-2 SUPER CONSTELLATION is rugged and for many of Wright Turbo-Compound engines.

Turbo-Compound Super Connie Debut

Lockheed Aircraft Corp. has rolled out its first Wright Turbo-Compound powered Super Constellation, which, according to company officials, will give the 12 foreign and domestic airlines that have ordered 69 of the transports a speed and long-range performance combination "undeniable even to today's jets."

Even superficial comparisons seem to put the Burbank, Calif., transport builder on a par ground for these up-transport charts since de Havilland hardly will claim that their present Constellation can match the new Super Constellation's ability to fly a huge number of passengers over longer, non-stop distances in one-day capability over the Atlantic, for example, from the Turbo-

Compound Super Connie definite speed and load advantages over the Lockheed Constellation 1 and 2, which are not true fast Atlantic passenger aircraft.

The 1400 Super Connie will have a top speed approaching 400 mph, with cruising speeds on long legs in the neighborhood of 340 mph. Engines are designated 9721C10D1A and have a rated output of 3,238 hp (dry). Recent Navy tests on a military Turbo-Compound engine showed a rating of 3,700 hp, with water injection, indicating the engine's capability for dual development.

The first Turbo-Compound Super Connie scheduled to make its maiden flight this week, is going to Navy and

carry the daughter RTV-1 in carry 101 passengers, up to 12 tons of cargo or 75 vehicles, etc. The nose is lifted with a large, ball, radome housing out of doors.

Another RTV-1 is undergoing static tests in a specially built \$60,000 test rig, where powerful hydraulic jacks have been applying multiple loads on the structure. No permanent distortion is reported as a result of these tests.

A later Navy Super Connie, the RTV-3, will have 160WA T38 turbo-

Compound operation will begin to take delivery of the Turbo-Compound 1400 in January 1953, with KLM Royal Dutch Airlines and Air France among the initial customers. By spring

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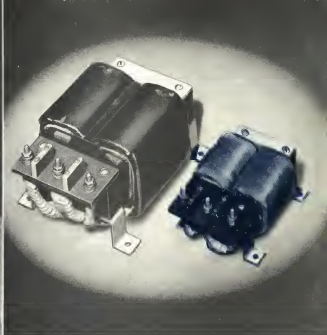
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AVIATION WEEK, November 3, 1952

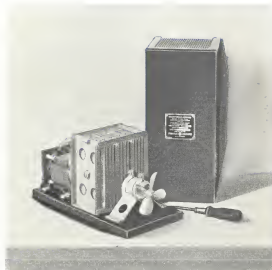


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